

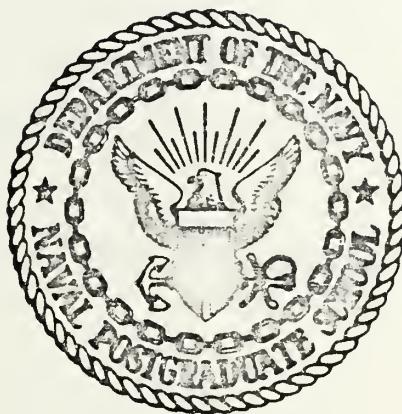
THE SINGULAR SURFACE IN THE PARAMETER
SPACE; APPLICATIONS

Paul Franklin Van Tassel

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THESIS

THE SINGULAR SURFACE IN THE PARAMETER
SPACE; APPLICATIONS

by

Paul Franklin Van Tassel

December 1975

Thesis Advisor:

G. J. Thaler

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The singular surface, a hypersurface of singular value in the parameter plane, is introduced.

Compensation techniques using the singular line are demonstrated and a computer program for determining the existance of singular lines is presented.

The Singular Surface in the Parameter Space; Applications

by

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Lieutenant, United States Navy
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Submitted in partial fulfillment of the
requirements for the degree of

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I. INTRODUCTION

Compensation of linear control systems by the classical methods of root locus and frequency response is adequate when there is a single variable. However when there are two or more variables compensation of physical systems for desired system dynamics becomes a complex task [1].

In modern control engineering systems characteristic equations with two or more unknown parameters result. In 1959, Mitrovic's Method [2] was developed to portray the relationship between the roots of the characteristic equation and two coefficients of the system. This method allowed the two lowest ordered coefficients of the characteristic equation to be varied. Siljak expanded this method to the Coefficient Plane to allow the variation of any two of the coefficients of the characteristic equation. For example:

$$f(s) = B_i s^i + B_j s^j + \sum_{\substack{k=0 \\ k \neq i, j}}^n a_k s^k \quad (1-1)$$

where: a_k are real constants

B_i and B_j are real variables

Application of this method resulted in constant zeta, omega, and sigma curves on the $B_i - B_j$ plane specifying the roots of equation (1-1) for any choice of ζ and ω_n .

In 1964, Siljak [3] extended Mitrovic's work so that

the two variable parameters may appear linearly in the coefficients of the characteristic equation. Consider the linear case of the characteristic equation:

$$f(s) = \sum_{k=0}^n a_k s^k = 0 \quad (1-2)$$

where: $a_k = b\alpha_k + c\beta_k + d_k$ (1-3)

and: and are real variable parameters

Application of this method resulted in constant zeta, omega and sigma curves for any choice of α and β on the $\alpha-\beta$ plane specifying the roots of equation (1-2).

Siljak's Parameter Plane Method provides the designer with a simple procedure for factoring a characteristic equation and displaying the results in a parameter plane diagram. It allows the designer to obtain information about system stability and the affect of parameter variations and adjustments on stability. In the general case the Parameter Plane Method can be applied to any control problem in which it is necessary to determine how the variation of one or more parameters effect the root location of a system.

In 1967, Bowie [4] discovered that a limitation existed in parameter plane theory. While attempting to solve a sixth order characteristic equation, Bowie found that he could not find a complete set of roots in an area of the parameter plane that should have contained only real roots. By substituting the values of α and β into the characteristic equation he found the roots. It was discovered that a set of complex roots existed corresponding to a $\zeta-\omega_n$ pair in the left-half s -plane. Further

investigation showed that this ζ - ω_n pair formed a straight line in the parameter plane. This was contrary to the then existing parameter plane theory which stated that a ζ - ω_n pair will only occupy a point in the parameter plane.

Bowie called this line of constant ζ -constant ω_n the SINGULAR LINE as it was formed by a singular value of ζ and ω_n . The singular line added a new dimension to the parameter plane because it allows the designer to hold one root location constant while moving others to achieve desired system characteristics by varying α and β .

II. THE SINGULAR LINE IN THE PARAMETER PLANE

A. THE PARAMETER PLANE

The Parameter Plane Method was developed by Siljak [3] as an extension to Mitrovic's Coefficient Plane. The parameter plane is a mapping of points in the s -plane into the $\alpha - \beta$ plane. The following is the mathematical development of that transformation.

Consider the characteristic equation:

$$f(s) = \sum_{k=0}^n a_k s^k \quad (2-1)$$

where: a_k ($k=0, 1, 2, \dots, n$) are real variables

and: $s = -\sigma + j\omega = -\zeta \omega_n \pm j\omega_n \sqrt{1-\zeta^2}$ (2-2)

Examination of figure 2.1 shows

$$-\zeta = \cos \theta \quad (2-3)$$

and $\sqrt{1 - \zeta^2} = \sin \theta$

By substituting equation (2-3), equation (2-2) may be expressed as:

$$s = \omega_n (\cos \theta + j \sin \theta) \quad (2-4)$$

and $s^k = \omega_n^k (\cos k\theta + j \sin k\theta)$ (2-5)

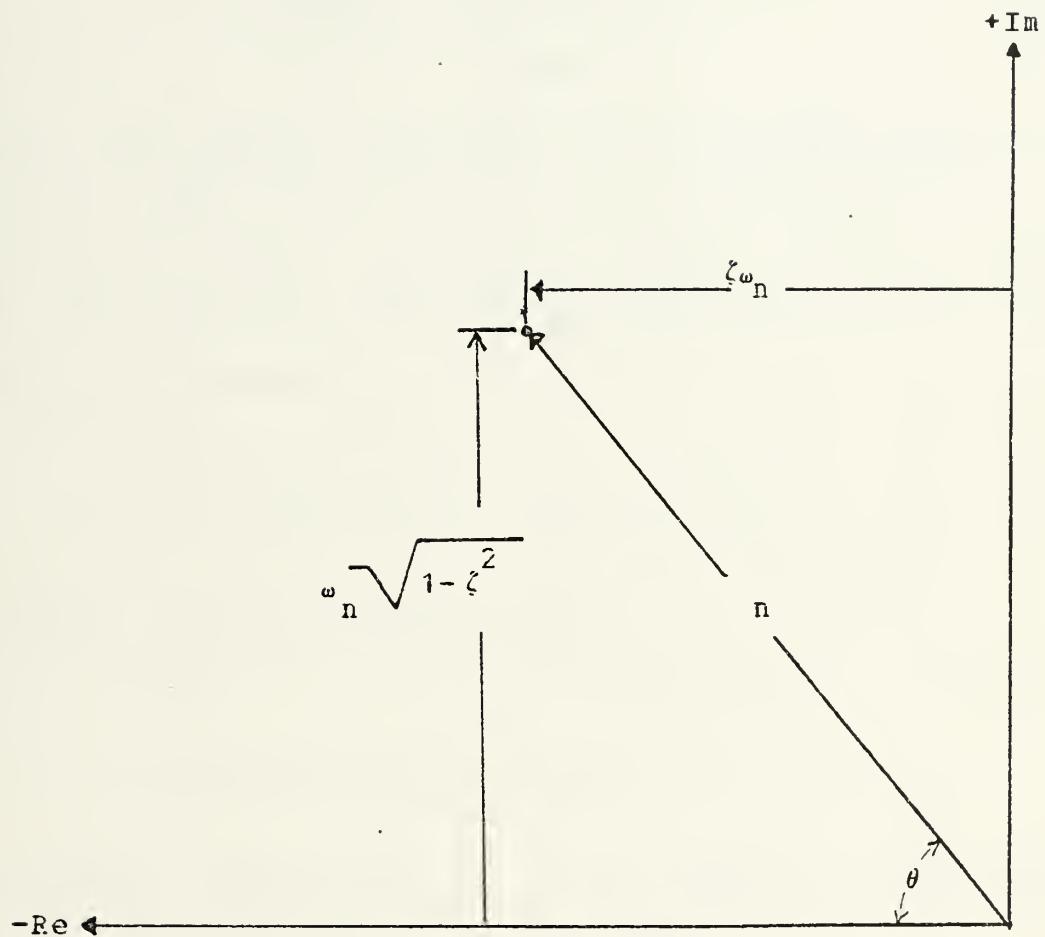


Figure 2.1
A POINT IN THE S-PLANE

Since

$$\theta = \cos^{-1}(-\zeta) \quad (2-6)$$

it follows that

$$\cos k\theta = \cos k[\cos(-\zeta)] \quad (2-7)$$

and

$$\sin k\theta = \sin k[\cos(-\zeta)] \quad (2-8)$$

Equation (2-7) can be expressed as the Chebyshev function of the first kind, $T_k(-\zeta)$ and similarly equation (2-8) can be expressed in terms of the Chebyshev function of the second kind, $\sin U_k[-\zeta]$. Substituting equations (2-7) and (2-8) equation (2-5) becomes

$$s^k = \omega_n^k [T_k(-\zeta) + j \sin U_k(-\zeta)] \quad (2-9)$$

where: $T_k(-\zeta) \zeta = (-1)^k T_k(\zeta)$

$$(2-10)$$

and $U_k(-\zeta) = (-1)^{k+1} U_k(\zeta)$

Substituting equation (2-10) into equation (2-9), then substituting equation (2-9) into equation (2-1), equation (2-1) becomes

$$f(s) = \sum_{k=0}^n a_k \omega_n^k (-1)^k T_k(\zeta) + j \sin \sum_{k=0}^n a_k \omega_n^k (-1)^{k+1} U_k(\zeta) = 0 \quad (2-11)$$

For any non-zero value of both the real and imaginary parts of equation (2-11) must go to zero independently therefore,

$$\sum_{k=0}^n a_k \omega_n^k (-1)^k T_k(\xi) = 0 \quad (2-12)$$

$$\sum_{k=0}^n a_k \omega_n^k (-1)^{k+1} U_k(\xi) = 0$$

The Chebyshev functions of the first and second kind are given by the recursion formulae

$$T_{k+1}(\xi) = 2T_k(\xi) - T_{k-1}(\xi) \quad (2-13)$$

$$U_{k+1}(\xi) = 2U_k(\xi) - U_{k-1}(\xi)$$

where:

$T_0(\xi) = 1$	$U_{-1}(\xi) = -1$
$T_1(\xi) = \xi$	$U_0(\xi) = 0$
	$U_1(\xi) = 1$

To eliminate $T_k(\xi)$ from equation (2-12) it is noted that

$$T_k(\xi) = U_k(\xi) - U_{k-1}(\xi) \quad (2-14)$$

then equation (2-12) may be rewritten

$$\sum_{k=0}^n (-1)^k a_k \omega_n^k [U_k(\xi)] - \sum_{k=0}^n (-1)^k a_k \omega_n^k U_{k-1}(\xi) = 0 \quad (2-15a)$$

$$\sum_{k=0}^n (-1)^{k+1} a_k \omega_n^k U_k(\xi) = 0 \quad (2-15b)$$

Examination of equation (2-15a) shows that its first part is

a linearly dependent solution to equation (2-15b) for all non-zero values of zeta therefore equation (2-15) becomes

$$\sum_{k=0}^n (-1)^k a_k \frac{u_{k-1}}{n} (\zeta) = 0 \quad (2-16)$$

$$\sum_{k=0}^n (-1)^k a_k \frac{u_k}{n} (\zeta) = 0$$

For the linear case:

$$a_k = \alpha b_k + \beta c_k + d_k \quad (2-17)$$

where: b_k , c_k , and d_k are real constants

and: α and β are the variable parameters

Substitution of equation (2-17) into equation (2-16) yields:

$$B_1 \alpha + C_1 \beta + D_1 = 0 \quad (2-18)$$

and

$$B_2 \alpha + C_2 \beta + D_2 = 0$$

where:

$$B_1 = \sum_{k=0}^n (-1)^k b_k \frac{u_{k-1}}{n} (\zeta) \quad B_2 = \sum_{k=0}^n (-1)^k b_k \frac{u_k}{n} (\zeta)$$

$$C_1 = \sum_{k=0}^n (-1)^k c_k \frac{u_{k-1}}{n} (\zeta) \quad C_2 = \sum_{k=0}^n (-1)^k c_k \frac{u_k}{n} (\zeta) \quad (2-19)$$

$$D_1 = \sum_{k=0}^n (-1)^k d_k n_{k-1}^k U_{k-1}(\xi) , \quad D_2 = \sum_{k=0}^n (-1)^k d_k n_k^k U_k(\xi)$$

Application of Cramer's Rule for the solution of simultaneous linear equations to equation (2-18) yields the desired parameter plane solution equation:

$$\alpha = \frac{C_1 D_2 - C_2 D_1}{B_1 C_2 - B_2 C_1} \quad \beta = \frac{B_2 D_1 - B_1 D_2}{B_1 C_2 - B_2 C_1} \quad (2-20)$$

Siljak [3] defined the parameter plane as a rectangular plot with α as the abscissa and β as the ordinate. Equation (2-20) gives α and β as a function of ω_n and ξ . By fixing $\omega_i = \omega_i$ and varying ω_n over the range from zero to infinity in equation (2-20), a curve is produced in the $\alpha-\beta$ plane for the fixed value of zeta. This curve specifies the α and β pairs which will cause equation (2-1) to have a pair of roots corresponding to the fixed value of zeta. Similarly by fixing $\omega_n = \omega_{ni}$ and varying zeta from -1 to +1 in equation (2-20), a curve is produced in the $\alpha-\beta$ plane for the fixed value of ω_n . This curve specifies the α and β pairs which cause equation (2-1) to have a pair of roots corresponding to the fixed value of ω_n . Thus equation (2-20) allows the mapping of complex points in the s-plane to the parameter plane.

To transform the real axis points in the s-plane into the parameter plane, replaces s in equation (2-1) by $-\sigma$ to obtain:

$$f(s) = \sum_{k=0}^n a_k (-\sigma)^k = 0 \quad (2-21)$$

Now equation (2-17) is substituted into equation (2-21) to obtain:

$$\sum_{k=0}^n (b_k + \beta c_k + d) (-\sigma)^k = 0 \quad (2-22)$$

Simplifying equation (2-22) obtains:

$$B(\sigma) + \beta C(\sigma) + D(\sigma) = 0 \quad (2-23)$$

where:

$$B(\sigma) = \sum_{k=0}^n (-1)^k b_k \sigma^k$$

$$C(\sigma) = \sum_{k=0}^n (-1)^k c_k \sigma^k \quad (2-24)$$

$$D(\sigma) = \sum_{k=0}^n (-1)^k d_k \sigma^k$$

For a given value of σ the functions $B(\sigma)$, $C(\sigma)$, and $D(\sigma)$ are constants and equation (2-20) results in a straight line in the parameter plane with a locus of point corresponding to the real roots $s=-\sigma$.

A digital computer program PARAM A was written by Nutting [5] to aid in solving the characteristic equation and plotting the constant zeta, omega, found at the end of this thesis and a brief explanation is found in Appendix A.

EXAMPLE I:

Consider a system whose characteristic equation is the fourth order polynomial:

$$f(s) = s^4 + 8.5s^3 + (5\alpha + 28)s^2 + (12.5\alpha + 25\beta + 42.5)s + 50\beta + 25 = 0 \quad 2-25)$$

Figure 2.2 shows the parameter plane for the following values of zeta, omega, and sigma:

<u>OMEGA</u>	<u>ZETA</u>	<u>SIGMA</u>
3.0	0.00	0.5
4.0	0.30	1.0
5.0	0.35	2.0
6.0	0.40	3.0
7.0	0.45	4.0
8.0	0.50	5.0
9.0	1.00	6.0

Examination of figure 2.2 shows the region of stability bounded by $\Im=0.0$, $\Im=1.0$, and $\sigma=0.0$. All values of α and β inside this region will result in a stable system. The roots of the system can be determined by selecting a point in the plane and reading the corresponding values of zeta, omega, and sigma. Zeta and omega will form a complex of roots so that:

$$s = -j\omega_n \pm j\omega_n \sqrt{1-\Im^2}$$

and the sigma point will form the real roots of the form:

$$s = -\sigma$$

Although it may not be immediately obvious, there are two sigma points through each point in the region. Careful interpolation or the plotting of additional lines is required

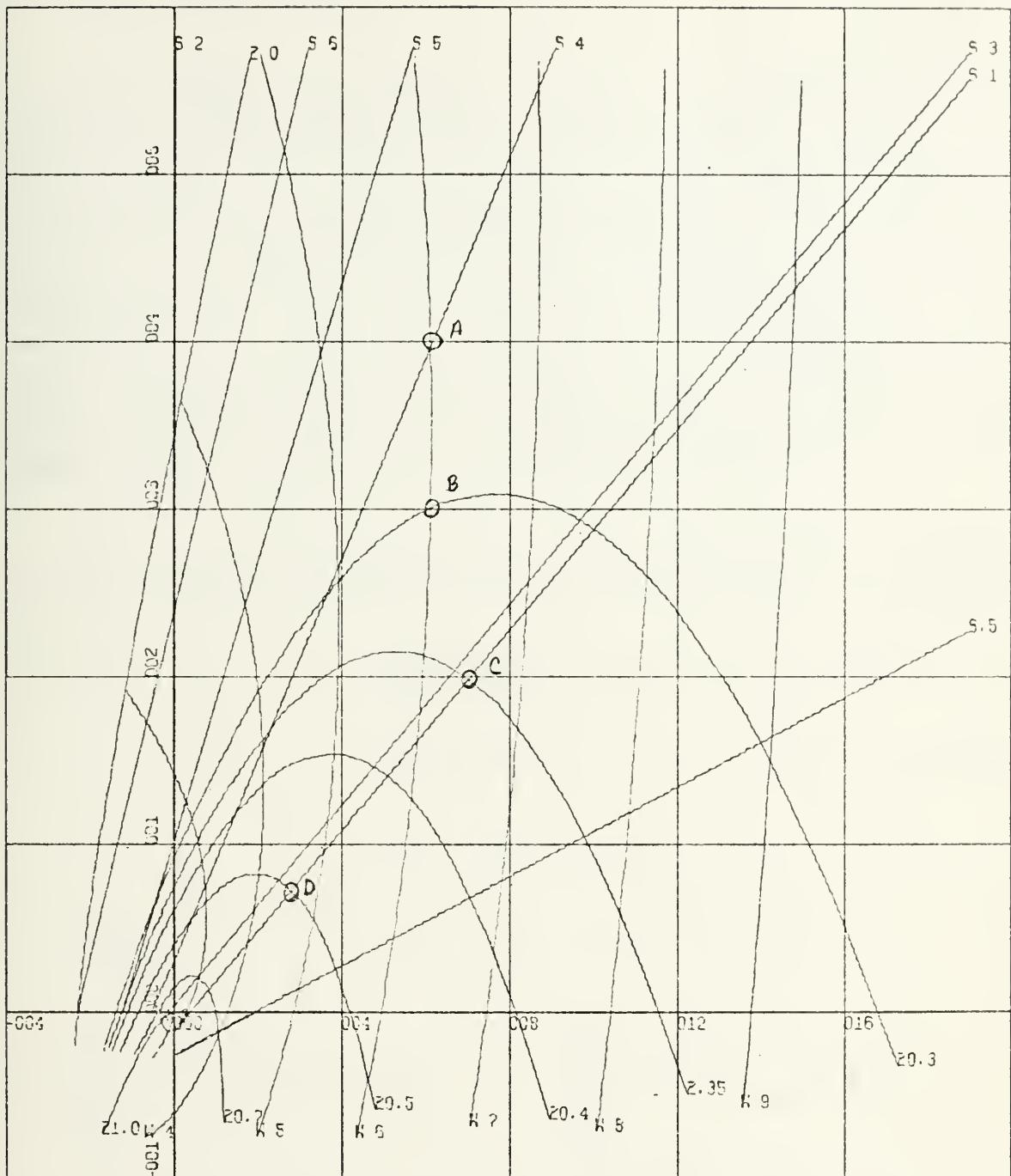
to obtain the exact roots of the system. But this is generally unnecessary as the designer will be interested in the general location of a specific pair of roots.

Four points have been selected from figure 2.2. The position of these points was measured with an engineer's scale and the resulting value of and were substituted in equation (2-25). The roots of the equation were then solved by the use of a polynomial rootfinder program on the IBM 360/67. Table II-1 shows the coordinates of the points, the resulting roots and the corresponding values of zeta, omega, and sigma. As can be seen the parameter plane provides a simple means of determining a root location while variable parameters are adjusted.

TABLE II-1

POINT	ALPHA	BETA	COMPLEX ROOTS	REAL ROOTS	OMEGA	ZETA
A	6.1	4.0	$-1.461 \pm j5.814$	-1.557 -4.020	5.995	.244
B	6.1	3.0	$-1.815 \pm j5.783$	-1.076 -3.794	6.061	.299
C	7.0	1.975	$-2.278 \pm j6.075$	-0.998 -2.945	6.490	.351
D	2.8	0.725	$-2.308 \pm j3.982$	-1.004 -2.881	4.603	.501

POINTS FROM FIGURE 2.2



α -scale: 4 units/inch
 β -scale: 1 unit/inch

Figure 2.2

PARAMETER PLANE EXAMPLE
 FOURTH ORDER POLYNOMIAL

B. THE SINGULAR LINE

As stated in Chapter I Bowie [4] discovered the existence of the singular line in the parameter plane. In investigating the cause of singular lines Bowie found that all previous work in the parameter plane was correct for as far as it went. It did not, however, give a full explanation of the case when both the numerator and denominator of equation (2-20) formed singular matrices. Siljak [3] did address the fact that this condition could exist, but the idea of a singular line in the parameter plane had not, up to that time, been conceived.

Bowie found that in order for singular lines to exist in the parameter plane the following conditions must be met:

$$\begin{aligned} B_1 C_2 - B_2 C_1 &= 0 \\ C_1 D_2 - C_2 D_1 &= 0 \\ B_2 D_1 - B_1 D_2 &= 0 \end{aligned} \quad (2-26)$$

where:

$$\begin{aligned} B_1 &= \sum_{k=0}^n (-1)^k b_k \sum_{ns}^k u_{k-1} (\gamma_s) & B_2 &= \sum_{k=0}^n (-1)^k b_k \sum_{ns}^k u_k (\gamma_s) \\ C_1 &= \sum_{k=0}^n (-1)^k c_k \sum_{ns}^k u_{k-1} (\gamma_s) & C_2 &= \sum_{k=0}^n (-1)^k c_k \sum_{ns}^k u_k (\gamma_s) \\ D_1 &= \sum_{k=0}^n (-1)^k d_k \sum_{ns}^k u_{k-1} (\gamma_s) & D_2 &= \sum_{k=0}^n (-1)^k d_k \sum_{ns}^k u_k (\gamma_s) \end{aligned} \quad (2-27)$$

and: ω_{ns} and ζ_s are the constant ω -constant ζ_s values.

equations (2-18) are solved for β in terms of ζ the
it is two linearly independent equations:

$$\beta = -\frac{D_1}{C_1} - \frac{B_1}{C_1} \alpha = -\frac{D_2}{C_2} - \frac{B_2}{C_2} \alpha \quad (2-28)$$

Equation (2-28) is the singular line solution. It
is a straight line in the parameter plane with a slope of
 C_1 and it intercepts the α -axis at $-D_1/C_1$. All points on
1

line represent a singular value of zeta and a singular
value of omega and therefore map into a single pair of
complex points in the s-plane where:

$$s = -\zeta_s \omega_{ns} \pm j\omega_{ns} \sqrt{1-\zeta_s^2} \quad (2-29)$$

It is obvious that singular lines do not exist for all
characteristic equations and when they do exist they will
exist only for the s - ω pairs which will satisfy equation
(26). To aid in determining the existence of singular
lines and plotting them in the parameter plane the computer
program SINGULAR LINE has been written.

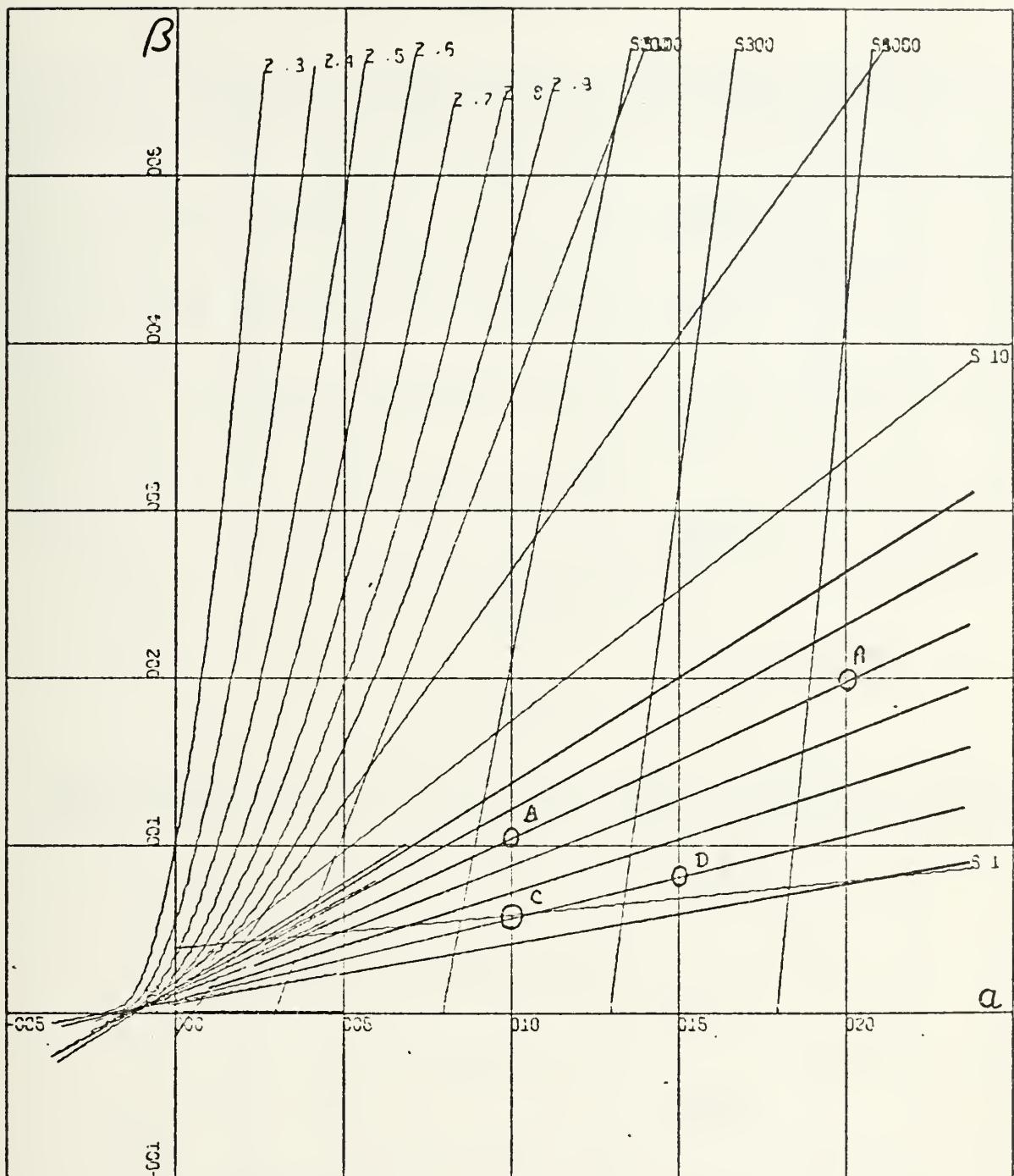
Briefly, the program uses the real constants of the
coefficients of the characteristic equation and the desired
value of ζ_s to solve for value for any real positive values
 ω_{ns} in equation (2-26). The ζ_s and ω_{ns} values are then
used to generate equations (2-27) and these results are used
to test for the existence of singular lines. If singular

region there will be two real roots and a pair of complex roots corresponding to the constant zeta-constant omega values of the singular lines.

A pair of points has been selected along two of the singular lines. Points A and B are on the line for $\zeta=0.7$, $\omega_n = 13.28$ and point C and D are on the line for $\zeta=0.4$ $\omega_n = 12.309$. Table II-2 shows the coordinates, the resulting roots and the values of zeta and omega. For Table II-2 the coordinates of the point were measured with an engineer's scale and the values of alpha and beta were substituted into equation (2-30) and the roots were computed using a polynomial rootfinder subroutine on the IBM 360/67.

Table II-3 is for the same points, the only difference being that the coordinates were computed using the slope-intercept method utilizing the printed output from the program SINGULAR LINE.

Table II-3 shows that this method of obtaining system roots provides an accurate method for obtaining singular lines and that the lines a for constant zeta-constant omega values, while Table II-2 shows that reasonably accurate results can be obtained using a minimum of equipment.



α -scale: 5 units/inch
 β -scale: 1 unit/inch

Figure 2.3

SINGULAR LINES ON THE PARAMETER PLANE
 FOURTH ORDER POLYNOMIAL

TABLE II-2

POINT	ALPHA	BETA	COMPLEX ROOTS	REAL ROOTS	OMEGA	ZETA
A	20.	19.67	$-9.244 \pm j10.87$	-9.326 -422.6	13.40	.689
B	10.	10.5	$-9.281 \pm j9.670$	-1.051 -222.4	13.40	.693
C	10.	5.667	$-4.839 \pm j11.19$	-9.971 -231.3	12.20	.397
D	15.	8.167	$-4.918 \pm j11.27$	-1.045 -326.7	12.30	.400

POINTS FROM FIGURE 2.3

TABLE II-3

POINT	ALPHA	BETA	COMPLEX ROOTS	REAL ROOTS	OMEGA	ZETA
A	20.	19.86	$-9.298 \pm j9.486$	-1.071 -422.3	13.40	.700
B	10.	10.52	$-9.298 \pm j9.486$	-1.051 -222.4	13.40	.700
C	10.	5.755	$-4.920 \pm j11.27$	-9.982 -231.2	12.30	.400
D	15.	8.225	$-4.920 \pm j11.27$	-1.011 -331.2	12.30	.400

POINTS FROM FIGURE 2.3

III. PROPERTIES OF THE SINGULAR LINE

To fully understand the relationship of singular lines to the parameter plane it is necessary to examine what the singular line is and what it represents.

The singular line exists as a unique situation that occurs while mapping the s-plane into the parameter plane. The necessary condition that must exist is that the determinants of the matrices which form equation (2-20) must be singular. When this occurs, a discontinuity appears in the mapping from the s-plane to the parameter plane. The result of this discontinuity is the singular line.

The discontinuity occurs for the singular value of zeta and the singular value of omega, and it maps into the parameter plane as a straight line from plus infinity to minus infinity. The equation of this line is

$$\beta = -\frac{D_1}{C_1} - \frac{B_1}{C_1}\alpha = -\frac{D_2}{C_2} - \frac{B_2}{C_2}\alpha \quad (3-1)$$

The slope of the line is $-B_1/C_1$ and the beta intercept is $-D_1/C_1$.

Until this writing it was thought that if a value of alpha and beta from any point along the singular line was placed in the characteristic equation, the system would have a pair of complex roots corresponding to the value of the constant zeta-constant omega of the singular line. However it is shown in the following examples that this result does

not always occur.

EXAMPLE III

Consider the following aircraft pitch stabilization system [6].

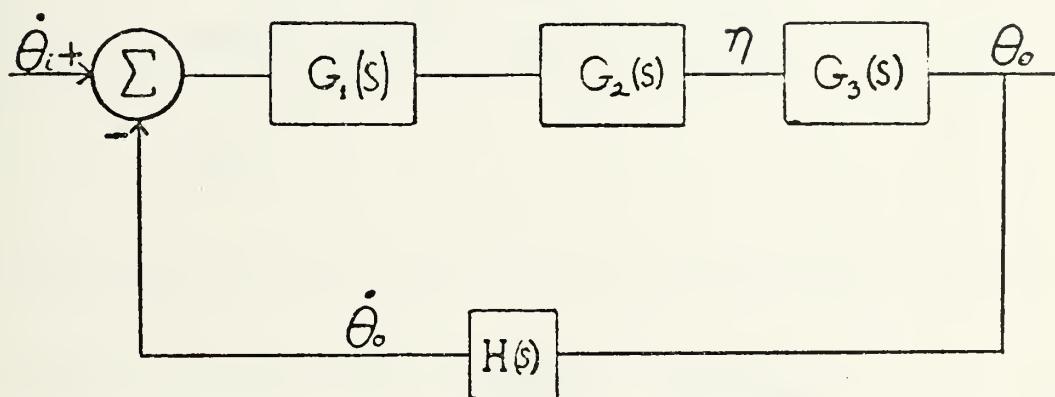


Figure 3.1
AIRCRAFT PITCH STABILIZATION SYSTEM

where: $\dot{\theta}_i$ is the pitch rate command

θ_o is the pitch rate

θ_o is the pitch angle

η is the elevator angle

and: $G_1(s)$ is a pole-zero compensator

$G_2(s)$ is a hydraulic servo

$G_3(s)$ is the airframe

$H(s)$ is a rate gyro

The transfer functions of the system components are:

$$G_1(s) = \frac{s+\alpha}{s+\beta}$$

$$G_2(s) = \frac{K_1 3180}{s^2 + 54.85 + 3201.01}$$

$$G_3(s) = \frac{K_2 (19.41)(s + 3.22)}{s(s + 7.58s + 124.614)}$$

$$H(s) = s$$

The characteristic equation for the uncompensated system is:

$$s^5 + 62.38s^4 + 3741.01s^3 + (31092.5 + 61732.6K)s^2 + (398891 + 198779K)s = 0 \quad (3-2)$$

$$\text{where } K = K_1 K_2$$

This system is unstable for $K > 3.27$. It is desired to compensate for $K = 5$ and to avoid the natural frequencies of the airframe, $\omega_n = 3.24$, and the hydraulic servo, $\omega_n = 7.049$

The characteristic equation of the compensated system is:

$$s^6 + (62.38 + \beta)s^5 + (3741.01 + 62.38\beta)s^4 + (31092.5 + 61732.8K + 3741.01\beta)s^3 + (398891 + 198751K\alpha + 31092.5\beta)s^2 + (198751K\alpha + 398891\beta)s = 0 \quad (3-3)$$

This characteristic equation with $K = 5$ was used with the program PARAM A to obtain the parameter plane curves, figure 3.2, and with the program SINGULAR LINE to determine

the existance of singular line. Two singular lines were found for each zeta value of 0.1, 0.2, 0.3, 0.4, and 0.5, and one singular line was found for each zeta value of 0.6, 0.7, 0.8, and 0.9.

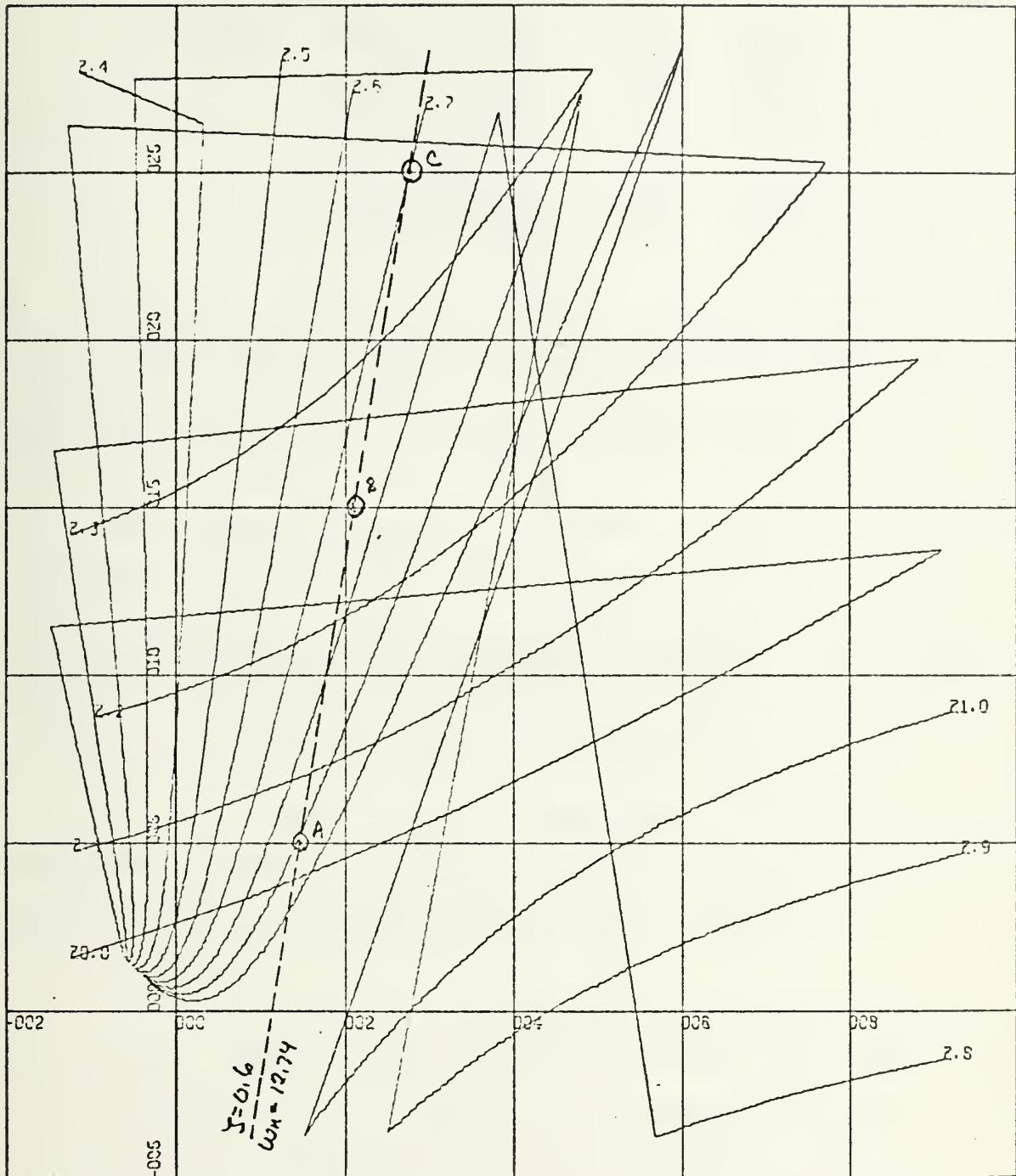
Zeta = 0.6 was chosen for compensation as it has an omega of 12.74 radians and would result in a fairly fast response without causing resonance at the natural frequencies of the airframe and the servo. The singular line for zeta = 0.6 is the broken line on the parameter plane of figure 3.2.

Three points were selected on the singular line. The points were; A(14.56,50), B(21.13,150) and C(27.69,250). These values of alpha and beta where put into the characteristic equation and the roots were solved for on the IBM 360 with a polynomial rootfinder routine.

There were two real roots and two pairs of complex roots for each of the points. Contrary to the supposition that all points along the singular line will produce a pair of complex roots corresponding to the value of the zeta-omega of that line, in this case such roots were not found. The singular line with zeta = 0.6 and omega = 12.74 should have according to prevoius concepts result in a complex root pair located at

$$s = -7.645 \pm j10.194 \quad (3-4)$$

This was not found to be the case. The complex roots that were found did not agree with the singular line roots, but they did agree with the parameter plane roots. These results are as follows:



α -scale: 20 units/inch
 β -scale: 50 units/inch

Figure 3.2

AIRCRAFT PITCH STABILIZATION SYSTEM PARAMETER PLANE WITH ZETA=0.6-OMEGA=12.74 SINGULAR LINE

<u>POINT</u>	<u>COMPLEX ROOTS</u>	<u>ZETA</u>	<u>OMEGA</u>
A	-8.083±j3.718 -2.512±j69.00	.908 .036	8.897 69.04
B	-7.944±j6.987 -16.66±j64.41	.751 .251	10.58 66.53
C	-7.871±j8.012 -20.75±j59.36	.701 .330	11.23 62.87

Identical results were found when values were taken from other singular lines in the same relative area of the parameter plane. All the roots were found to agree with the parameter plane, but not with the singular lines.

The initial response to these results was that the singular lines that were found were incorrect. However, after carefully checking the program siat were found were incorrect. However, after carefully checking the program SINGULAR LINE and hand calculating two of the singular lines, the program was deemed to be correct.

In an attempt to find a solution to this dilemma the singular line examples given by Bowie were examined. In all cases the singular lines were located outside the region of the normal complex root parameter plane curves.

In the belief that the constant zeta curves in figure 3.2 formed a loop as omega increased, values of alpha and beta believed to be outside the region were tried. The straight lines connecting the zeta curves in the figures shown in this thesis are the result of the DRAW routine and are not part of the parameter plane. They result when intermediate points are not in the area of the plot. Again three points on the same singular line were selected. These points were D(76.93,1000), E(142.57,2000) and F(208.22,3000). These points are not located on figure 3.2, but the location was computed using the slope-intercept

method. These results are as follows:

<u>POINT</u>	<u>COMPLEX ROOTS</u>	<u>ZETA</u>	<u>OMEGA</u>	
D	-7.725±j9.542 -23.31±j51.08	.629 .415	12.28 56.15	*
E	-7.688±j9.856 -23.47±j49.55	.615 .428	12.50 54.59	*
F	-7.675±j9.966 -23.50±j49.04	.610 .432	12.58 54.38	*

It is evident from the above that one pair of complex roots, indicated by the astrisk, is approaching the value of those which should result due to the singular line and that the value of zeta and omega for that pair of roots is relatively insensitive to large changes in alpha and beta.

In light of these results an examination of the parameter plane in the region of these points was conducted. For reasons of clarity only three values of zeta were plotted. These were 0.0, 0.6, and 1.0. The resulting parameter plane is seen in figure 3.3. As can be seen the constant zeta lines do not loop around as previously thought, but are discontinuous and the singular line approaches asymptotically.

What has happened is that as omega was increased on the constant zeta curve and approached the singular point the matrix which forms the denominator of equation (2-20) becomes very much smaller than the numerator causing the constant zeta line to go to plus infinity at the singular point asymptotic to the singular line. Once the value of omega is past the singular point it reenters the parameter plane at minus infinity, asymptotic to the singular line. Since there is only one singular point for this value of zeta the constant zeta line will continue through the parameter plane approaching infinity once more as omega

approaches infinity.

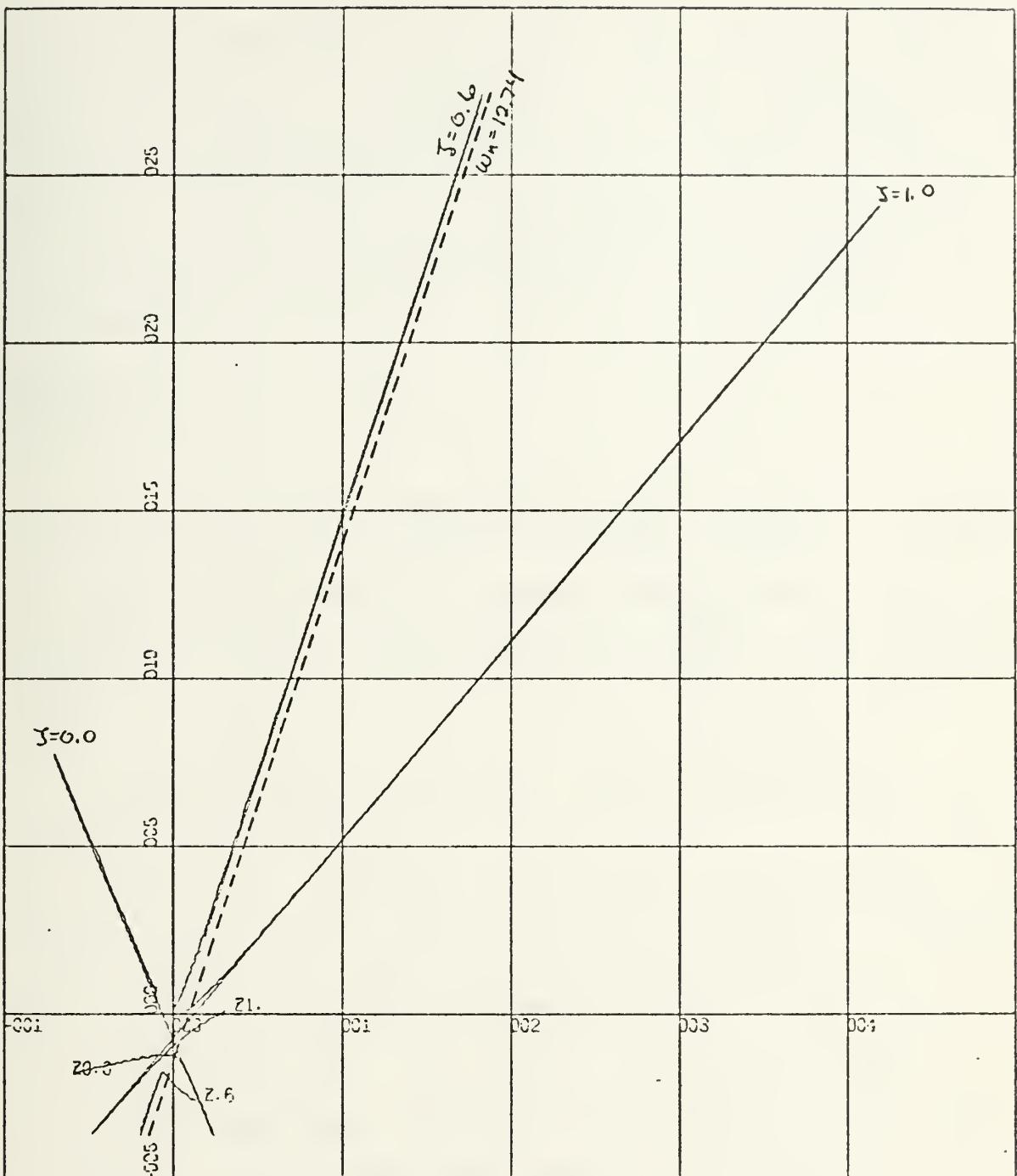
The reason that this occurs is evident when the mapping of the constant zeta line from the s-plane to the parameter plane is observed. The constant zeta line in the s-plane is continuous from $\omega = 0$ to $\omega = \infty$, therefore the mapping of this line into the parameter plane must also be continuous.

However there is a discontinuity at the singular point where the constant zeta curve in the parameter plane goes to infinity for a finite value of omega. The singular line provides a continuous mapping of the constant zeta curve as it traverses from plus infinity to minus infinity. This line exists at infinity, but not in the parameter plane as it is known.

In an effort to examine the change in alpha and beta as omega was increased a printout of PARAM A was made. However due to the iteration of omega in PARAM A it was not possible to obtain sufficient data around the singular point. In PARAM A for an omega of about 12 radians the iterative step is about 0.9 radians. Therefore the section of PARAM A which computes the constant zeta curves was modified to vary omega over 1 radian in steps of 0.001 radians starting 0.5 radians below the singular point and increasing to 0.5 radians above the singular point.

In order to see how small increases in omega affect the values of alpha and beta some of the points are shown:

<u>OMEGA</u>	<u>ALPHA</u>	<u>BETA</u>
12.242	88.56	1297.87
12.342	111.31	1644.01
12.442	149.21	2220.85
12.542	226.06	3391.11
12.642	451.58	6826.15
12.741	144447.	2200380.
12.743	-66473.4	-1012672.
12.842	-454.96	-6980.32
12.942	-230.44	-3564.49
13.042	-154.32	-2405.44



α -scale: 100 units/inch
 β -scale: 500 units/inch

Figure 3.3

PARAMETER PLANE FOR AIRCRAFT PITCH STABILIZATION SYSTEM
 $ZETA = 0.6$

As can be seen for large changes in alpha and beta in the area of the parameter plane where the singular line is asymptotic to the constant zeta curves results in small changes of omega. Since omega is very insensitive to changes on alpha and beta in this region the concept of the singular line is valid from the standpoint of the engineer. It is possible to vary a parameter along the singular line in this region and expect to maintain a constant value of zeta and omega.

EXAMPLE IV

In light of the unexpected results obtained in example III it was decided to carefully reexamine example II that was given in the previous chapter. The parameter plane was expanded around the area of the origin where the singular lines pass through the complex root portion of the parameter plane.

Figure 3.4 shows this area of the parameter plane. The constant zeta curves are for values of 0.0, 0.2, 0.5, 0.6, 0.8, and 1.0. There are two singular lines for each zeta value of 0.2, 0.5, 0.6, and 0.8 which are shown as dashed lines. There are two discontinuities in each zeta curve. Although it is not immediately obvious due to the limited area of the plot, all the constant zeta curves originate in the second quadrant, and become infinite at the first discontinuity and then reappear at infinity in the fourth quadrant. The curves, for values of zeta up to and including 0.5 then follow a path just to the left of the origin and enter the first quadrant where they again become discontinuous and infinite. The zeta curves for values of zeta greater than 0.5 come from infinity in the fourth quadrant and go to the left of the origin where they turn into the third quadrant where they again become

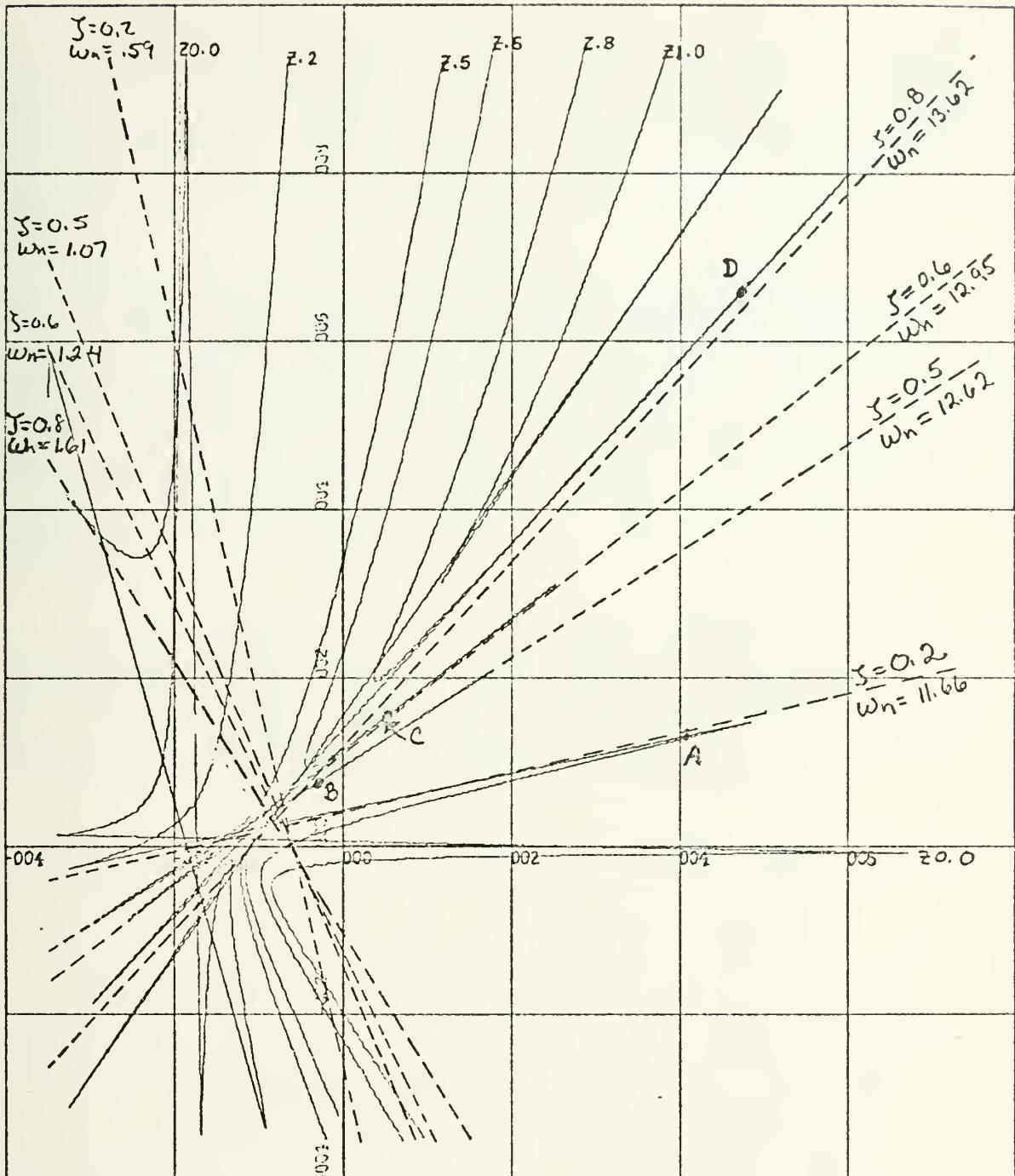
discontinuous and infinite. The curves for zeta values of 0.5 and less reappear at infinity in the third quadrant then go to infinity as a continuous function in the first quadrant. The curves for zeta values greater than 0.5 reappear at infinity in the first quadrant, sweep toward the origin then back through the first quadrant to infinity

The region of stability in the parameter plane is bounded by the zeta=0.0 curve, therefore the singular lines of interest lie in the first quadrant. As can be seen the zeta=0.5 and zeta=0.6 singular lines nearly coincide with the corresponding zeta curves in the parameter plane and the zeta=0.2 and zeta=0.8 lines are being asymptotically approached by the corresponding parameter plane zeta curves.

As in example III the modified section of PARAM A was used to obtain a printout of alpha and beta for a 1 radian change in omega as the constant zeta curve passed through the singular line value. In this example the constant zeta curves in the first quadrant approached their singular value very rapidly.

For purposes of illustration points have been placed on the constant zeta curves in figure 3.4 at the value at which omega is within 0.5 radians of the singular point which is at infinity. These points are:

<u>POINT</u>	<u>ZETA</u>	<u>ALPHA</u>	<u>BETA</u>
A	0.20	4.078	1.30
B	0.50	-0.542	0.59
C	0.60	0.571	1.60
C	0.80	4.754	6.62



-scale: 2 units/inch
 -scale: 2 units/inch

Figure 3.4

SINGULAR LINE THROUGH THE PARAMETER PLANE
 FOURTH ORDER POLYNOMIAL

As can be readily seen these points are within 5 percent of the singular value of omega and if one continues along the zeta curves it will be found, that for practical design purposes, the changes in omega are inconsequential.

It has been shown that a true singular line in the original concept does not exist. In examples III and IV it is shown that, if a designer carefully examines the singular lines and their relationship to the constant zeta curves of the parameter plane the singular concept can be used for practical design applications.

EXAMPLE V

This example is the original system in which Bowie found that singular lines existed in the parameter plane. The system which was used has a characteristic equation which is the following sixth order polynomial:

$$s^6 + 80s^5 + (20\alpha + 1600)s^4 + 840\alpha s^3 + (1600\alpha + 400\beta)s^2 + 1600\beta s + 1600\beta = 0 \quad (3-5)$$

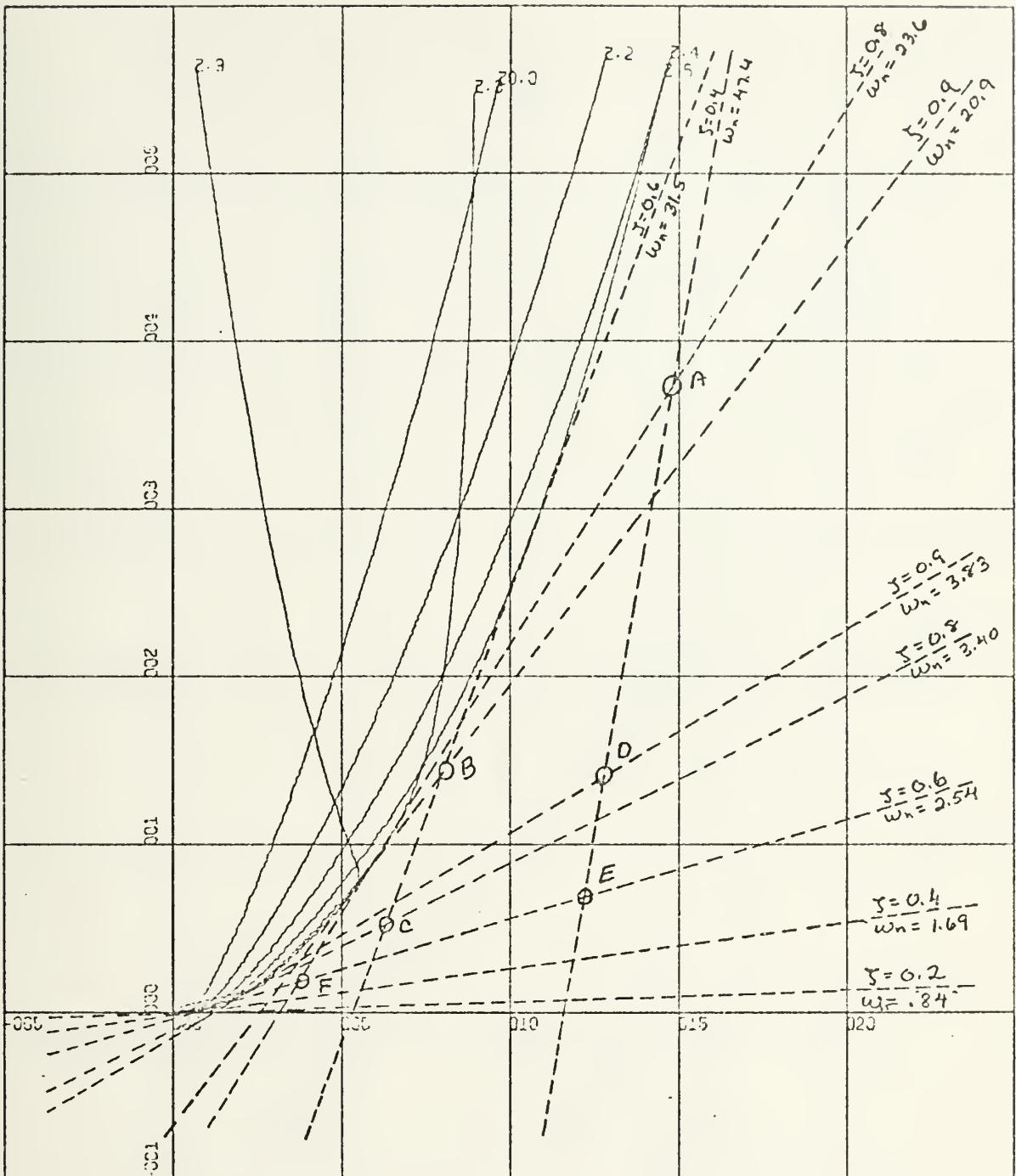
The parameter plane with constant zeta curves with the values of 0.0, 0.2, 0.4, 0.6, 0.8, and 0.9 is seen in figures 3.5 and 3.6. Figure 3.6 is the area around the origin in figure 3.5. There were two singular lines for each of the zeta values 0.2, 0.4, 0.6, 0.8, and 0.9 and these are shown on figures 3.5 and 3.6 as dashed lines.

As can be seen the parameter plane has no visible discontinuities as omega increases, however due to the fact that two singular lines exist for each value of zeta there must be two discontinuities in each zeta curve. Examination of the singular lines and the constant zeta curves of the

parameter plane indicates that the singular lines are tangent to the corresponding zeta curve at one point. When the values of zeta and omega for these points were examined it was found that the tangent points occur at the singular points in the constant zeta curves.

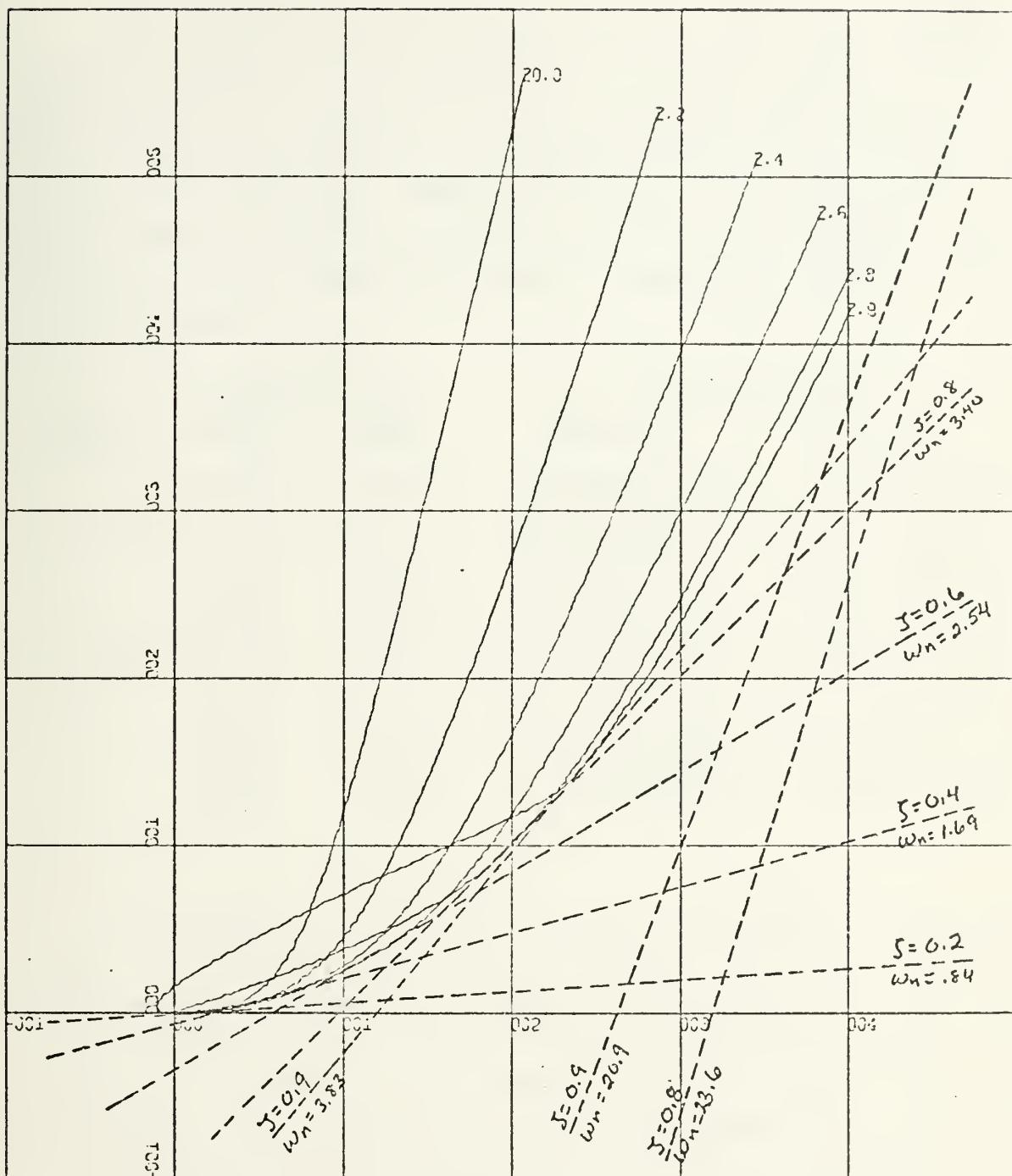
Unlike examples III and IV where the mapping from the s-plane approached infinity as the singular point was approached this example maps as though no discontinuity existed. When the discontinuity is reached the constant zeta curve immediately goes to plus infinity tangent to the curve at the point of discontinuity and returns from minus infinity tangent to the constant zeta curve and continues mapping to infinity as omega approaches infinity.

This phenomenon does not reveal itself in a normal parameter plane study for two reasons. The first of these is the previously mentioned size of iteration in the PARAM A program and the second reason is that a normal practice in computer programming is to limit the numerator of a term to a minimum size to prevent an overflow or underflow in the computer when a number beyond the computer's capability is reached. This practice was followed in the writing of PARAM A so if the singular point was reached on an iterative step the program would have recognized this as a forbidden number and continued to the next value of omega, thereby bypassing the singular point and continuing with no indication that this condition existed. If Bowie had not been interested in the area outside the normal complex root area of this example he would have had no indication that singular lines existed.



α -scale: 50 units/inch
 β -scale: 1000 units/inch

Figure 3.5
 SINGULAR LINE IN THE PARAMETER PLANE
 EXAMPLE V



α -scale: 10 units/inch
 β -scale: 100 units/inch

Figure 3.6

SINGULAR LINES IN THE PARAMETER PLANE
 EXAMPLE V - EXPANDED

In this example the singular lines are true singular lines as originally conceived. If a value of alpha and beta is selected at any point along these lines a pair of complex roots corresponding to the constant zeta-constant omega of that line will result. To show this fact six points have been selected in the region of singular line on figure 3.5. As can be seen by the following the expected complex roots were obtained.

POINT	SINGULAR POINT		COMPLEX ROOTS	
	ZETA	OMEGA	EXPECTED	OBTAINED
A	0.80	23.55	-18.44±j14.13	-18.42±j14.15
	0.40	47.41	-18.96±j43.15	-18.95±j43.44
B	0.90	20.88	-18.79±j9.101	-18.79±j9.121
	0.60	31.53	-18.92±j25.22	-18.91±j25.23
C	0.80	3.396	-2.717±j2.078	-2.721±j2.081
	0.60	31.53	-18.92±j25.22	-18.91±j25.22
D	0.90	3.831	-3.450±j0.728	-3.465±j0.729
	0.60	31.53	-18.92±j25.22	-18.90±j25.23
E	0.60	2.537	-1.522±j2.030	-1.524±j2.032
	0.40	47.41	-18.96±j43.45	-18.94±j43.43
F	0.60	2.537	-1.522±j2.030	-1.521±j2.029
	0.80	23.55	-18.44±j14.13	-18.47±j14.15

In summary, the singular lines result when a constant zeta or constant omega curve is mapped from the s-plane to the parameter plane and a discontinuity occurs. At this discontinuity the parameter plane mapping goes to infinity, either plus or minus, and returns from the other side of infinity. The singular line simply provides a mapping through this region to maintain a continuous mapping from the s-plane to the parameter plane. In cases, such as examples III and IV, where the mapping first approaches infinity as the singular point is approached the singular line concept is a valid engineering aid over a limited range of the singular line. However in cases such as example V the singular line may be used over its entire range as an

excellent design aid.

IV. THE SINGULAR SURFACE IN THE PARAMETER SPACE

A. PARAMETER SPACE

The parameter space is an extension of the parameter plane when there are three or more variable parameters [7], [8], [9]. Once the concept of the parameter plane is understood it is relatively easy to extend the concept to three or more variable parameters.

The easiest method to generate a three dimensional parameter space is to set the third variable parameter the third variable parameter to a known value and then generate the two dimensional parameter plane. The process is then repeated for additional values of the third parameter. The result is a series of parameter planes which will form a parameter space.

EXAMPLE VI

Consider the following system:

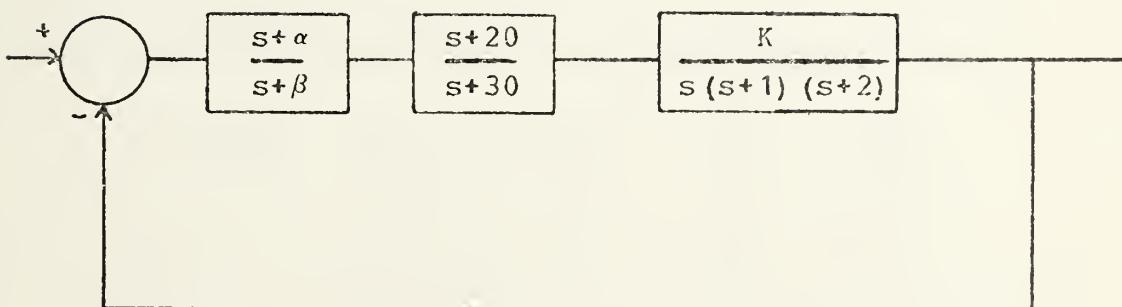


Figure 4.1

THIRD ORDER SYSTEM WITH DUAL CASCADE COMPENSATION

The characteristic equation for this system is:

$$s^5 + (33 + \beta)s^4 + (92 + 32\beta)s^3 + (50 + 92\beta + K)s^2 + (20K + \alpha K + 50)s + K\alpha = 0 \quad (4-1)$$

Figure 4.2 shows the parameter space for the constant $\zeta=0.5$ and constant $\omega=1.7$ and the values of K from 10 to 15. It was constructed by fixing the value of K and computing the parameter plane using PARAM A. The curves were then plotted onto a single plot to show the three dimensions.

Both the constant ζ curves and the constant ω curves form a surface and the interconnection of these surfaces forms a curved line through the parameter space. Any point on this line will generate a pair of complex roots corresponding to the s -plane point of $\zeta=0.5$ and $\omega=1.7$. For these values of ζ and ω the complex roots are:

$$s = -0.850 \pm j1.470 \quad (4-2)$$

By selecting three points on this line this is shown to be true. The points are A(0.16, 2.6, 10.0), B(0.347, 3.0, 12.0), and C(0.52, 3.63, 15.0). These points were taken from the parameter plane using an engineer's scale.

POINT	COMPLEX ROOTS	ZETA	OMEGA
A	-0.847±j1.478	.497	1.703
B	-0.849±j1.475	.498	1.701
C	-0.851±j1.469	.501	1.698

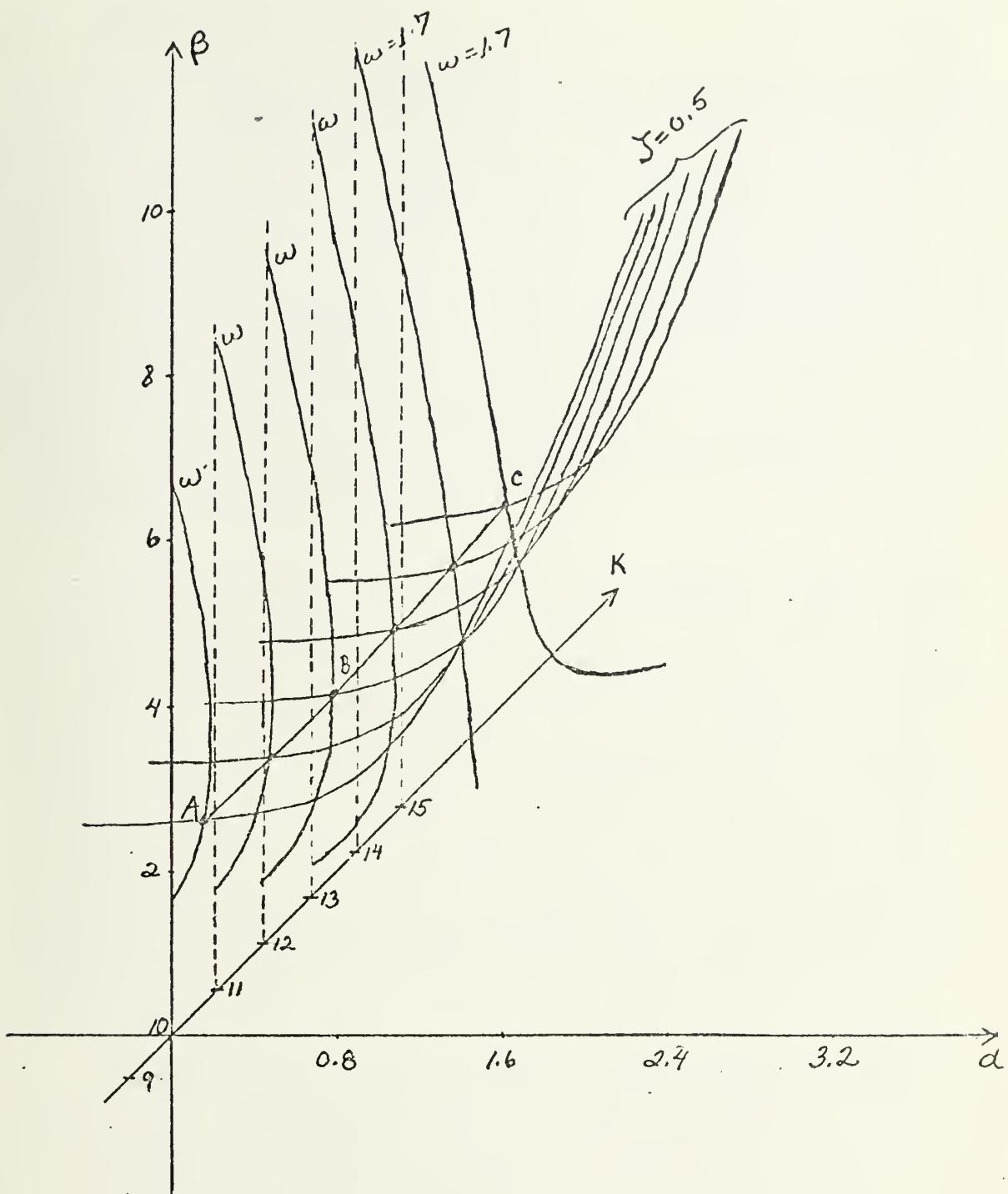


Figure 4.2

PARAMETER SPACE WITH ZETA=0.5 AND OMEGA=1.7 SURFACES

B. SINGULAR SURFACES

The parameter space is continuous throughout the range of the third variable parameter. If for any value of the third parameter a discontinuity appears in the parameter plane it will also appear in the parameter space as a discontinuity. As stated in chapter III this discontinuity will result in a singular line. As the third parameter is varied the discontinuity will result in singular lines for the same value of zeta and omega. The slope and the axis intercept points will change slightly so the result is a smooth surface representing a warped hypersurface traversing through the parameter space.

Figure 4.3 is a photograph of a singular surface model. The singular surface is from the system in example VI. the singular values are $\zeta=0.5$ and $\omega=12.04$ and the third variable K is varied from 5.0 to 20.0. As can be seen the surface is a smooth warped hyperplane. The model was constructed by obtaining the singular lines for the same singular value at each value of K , transferring them to cardboard, cutting them and accentting them with a contrasting color. The photography was done by the Photo Division, Education Media Department, Naval Postgraduate School.

Figure 4.4 is a drawing of the same singular surface as figure 4.3. The difference between the drawing and the photograph is that in the photograph the K axis starts at $K=5$ in the foreground continues back through to $K=20$, while in the drawing $K=5$ is the most distant singular line and the singular lines progress forward until the value of $K=20$ is reached.

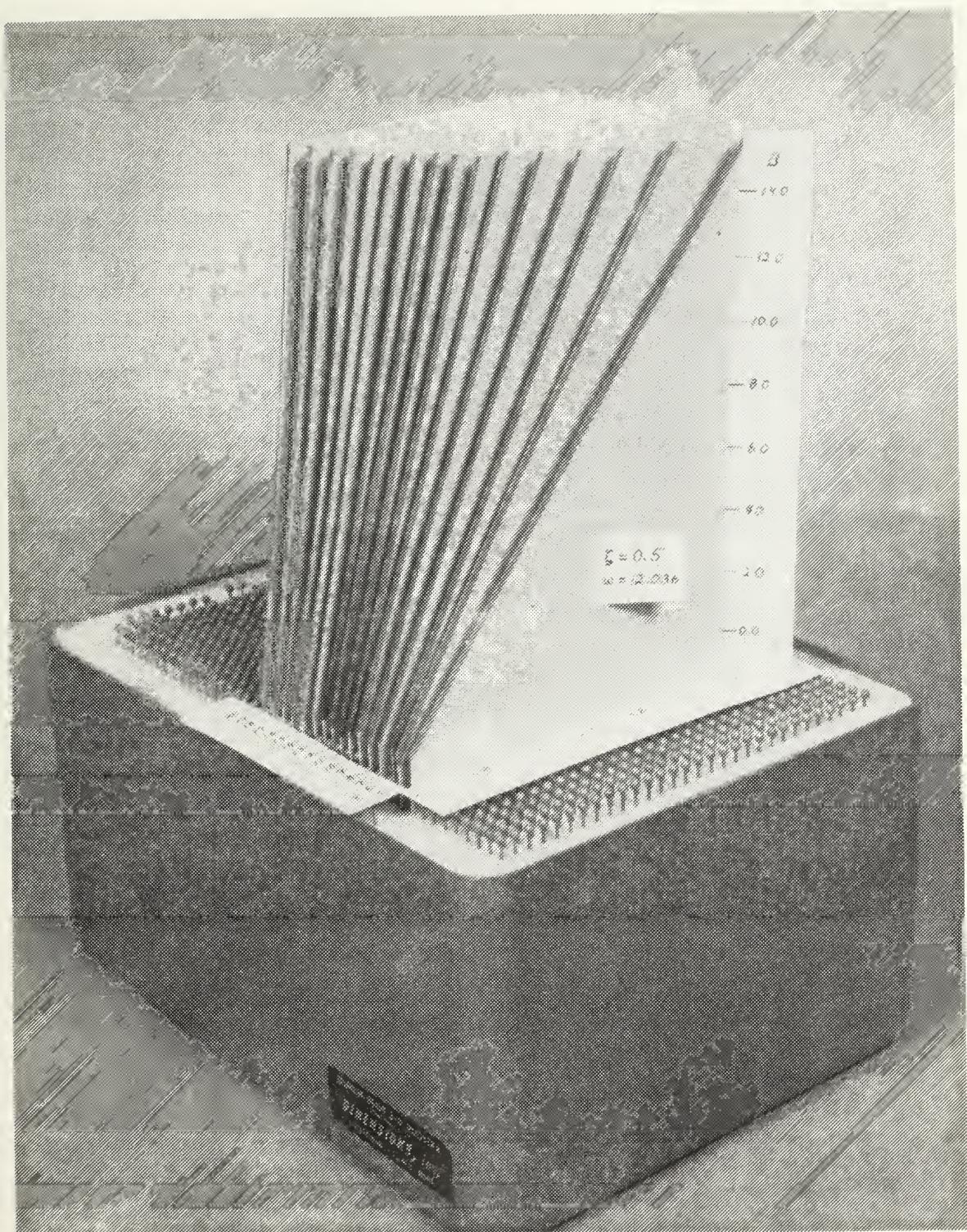


Figure 4.3
SINGULAR SURFACE MODEL

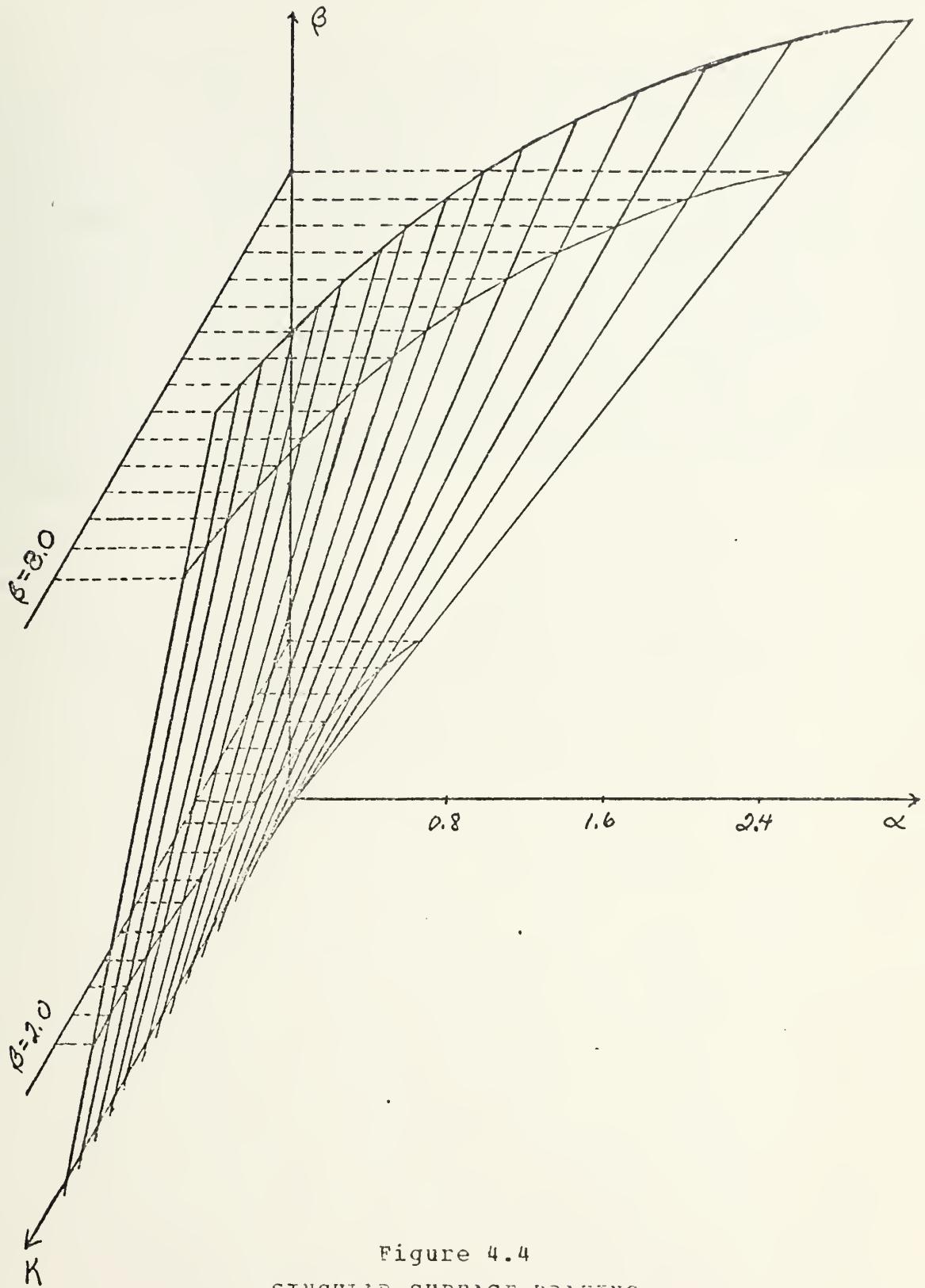


Figure 4.4
SINGULAR SURFACE DRAWING

The relationship of the singular surface to the parameter space is exactly the same as that of the singular line to the parameter plane. As with the singular line, the singular surface provides continuity at infinity. The designer must be aware of the type of discontinuity which is producing the surface. If the discontinuities occur like the ones in example III and IV then the principles of singular lines may only be applied in limited areas of the surface. If the singularities occur like those in example IV then any point on the singular surface will give a pair of complex roots corresponding to the constant zeta-constant omega of the surface.

V. COMPENSATION TECHNIQUES USING THE SINGULAR LINE

The major advantage of using singular lines for compensation is that as one moves up and down the singular line, within previously mentioned limits, one pair of complex roots remains in the same locations in the s-plane. While this pair of roots is stationary other root locations can be varied as the location on the singular line is moved through other singular lines, real root lines, and other constant zeta parameter plane curves.

There many excellent texts available which go into great detail on how root location will affect system response, overshoot, damping and so forth [1], [10]. Once the basic principles on how root locations affect a system are understood it is possible to use singular line concepts to obtain proper root locations.

EXAMPLE VIII

Consider the pitch stabilization system given in example III.

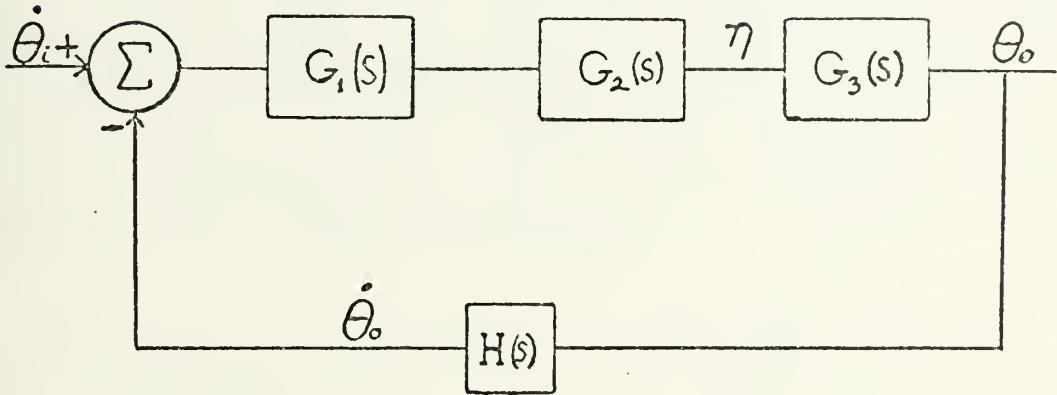


Figure 5.1
AIRCRAFT PITCH STABILIZATION SYSTEM

The characteristic equation for the system is:

$$\begin{aligned}
 & s^6 + (62.38 + \beta) s^5 + (3741.01 + + 62.38\beta) s^4 \\
 & + (31092.5 + 61732.8K + 3741.01\beta) s^3 + (398891 + 198751K\alpha \\
 & + 31092.5\beta) s^2 + (198751K\alpha + 398891\beta) s = 0
 \end{aligned} \tag{5-1}$$

It was shown in example III that singular lines exist for this system, but they are the result of discontinuities at infinity. Therefore in order to effectively use the singular line concept it is necessary to examine the singular lines to determine the areas in which singular results could be expected.

In the case of this problem it was decided to use the singular line with $\zeta=0.6$ and $\omega=12.74$ as a dominant

root pair to give fast system response and avoid the natural frequencies of the servo and the airframe. Figure 5.2 shows the singular lines for this system for values of zeta of 0.4, 0.5, 0.6, 0.7, 0.8, and 0.9.

By moving up the $\text{zeta}=0.6$, $\omega=12.74$ singular line above the region where $\beta=3000$ the value of zeta and omega should remain relatively constant. Two points have been selected to attempt compensation, A(273.87, 4000) and B(306.69, 4500). These values have been put into the characteristic equation and the system was simulated using a step input of 0.2 radians per second.

Figure 5.3 is the response for the point A values of alpha and beta and figure 5.4 is the response for the point B values of alpha and beta. As can be seen the system response is essentially the same for both points. The response from the point A values has a peak overshoot of 1.08 and the response from the point B values has a peak overshoot of 1.1. The steady state output is 1.66 degrees in both cases.

This demonstrates one of the main advantages of the singular line concept. As two parameters are varied the system response remains insensitive to the change.

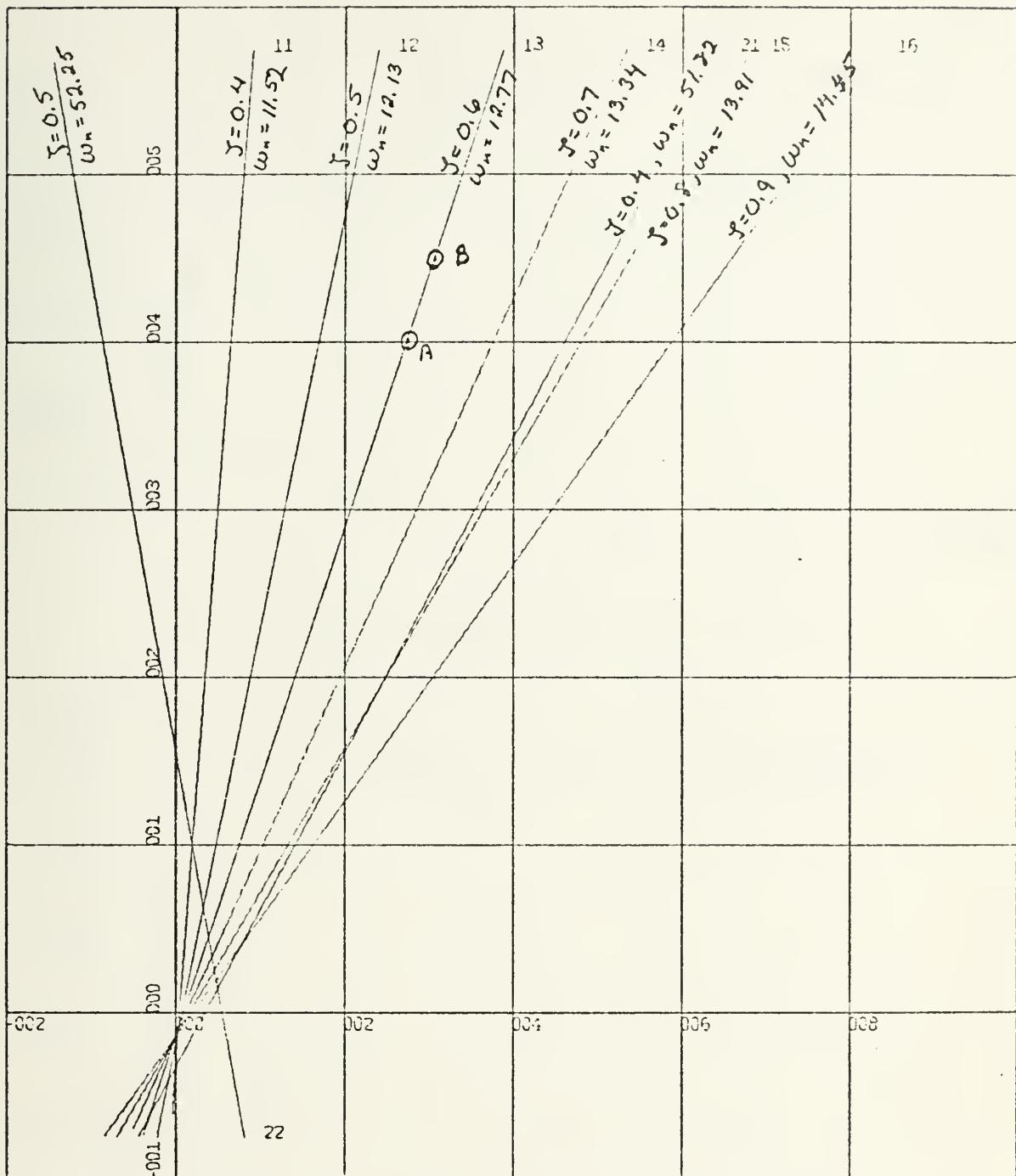
If this parameter variation were to take place inside the normal complex root region of the parameter plane the shift of root locations would result in a change of output response. Figure 5.5 is the parameter plane of the normal complex root area of the airdraft pitch stabilization system. Between point A and point C the linear change in beta is 200 and the change in alpha is 13.13.

There are two pairs of complex roots and two real roots in this section of the parameter plane. One of the real

roots is cancelled at the origin of the s-plane by the zero produced by the rate gyro in the feedback loop. The other real root location is past $s=-75$ in these cases and will only have a slight affect on the steady state value of the output. The complex root locations and their corresponding values of zeta and omega are:

<u>POINT</u>	<u>COMPLEX ROOTS</u>	<u>ZETA</u>	<u>OMEGA</u>
A	-8.083±j3.718 -2.512±j69.00	.908 .036	8.897 69.04
B	-7.944±j6.987 -16.66±j64.41	.751 .251	10.58 66.53
C	-7.871±j8.012 -20.75±j59.36	.701 .330	11.23 62.87

At point A the dominant root is located at $s = -2.512 \pm j69.00$. At point C the dominant root is located at $s = -7.871 \pm j8.012$. It can be seen that at the point A values the system response will be almost undamped with a peak overshoot of about 1.9 and a natural frequency of about 69 radians, while at the point C values the system will be well damped with a natural frequency of about 11 radians.



α -scale: 200 units/inch
 β -scale: 1000 units/inch

Figure 5.2
 SINGULAR LINES FOR
 AIRCRAFT PITCH STABILIZATION SYSTEM

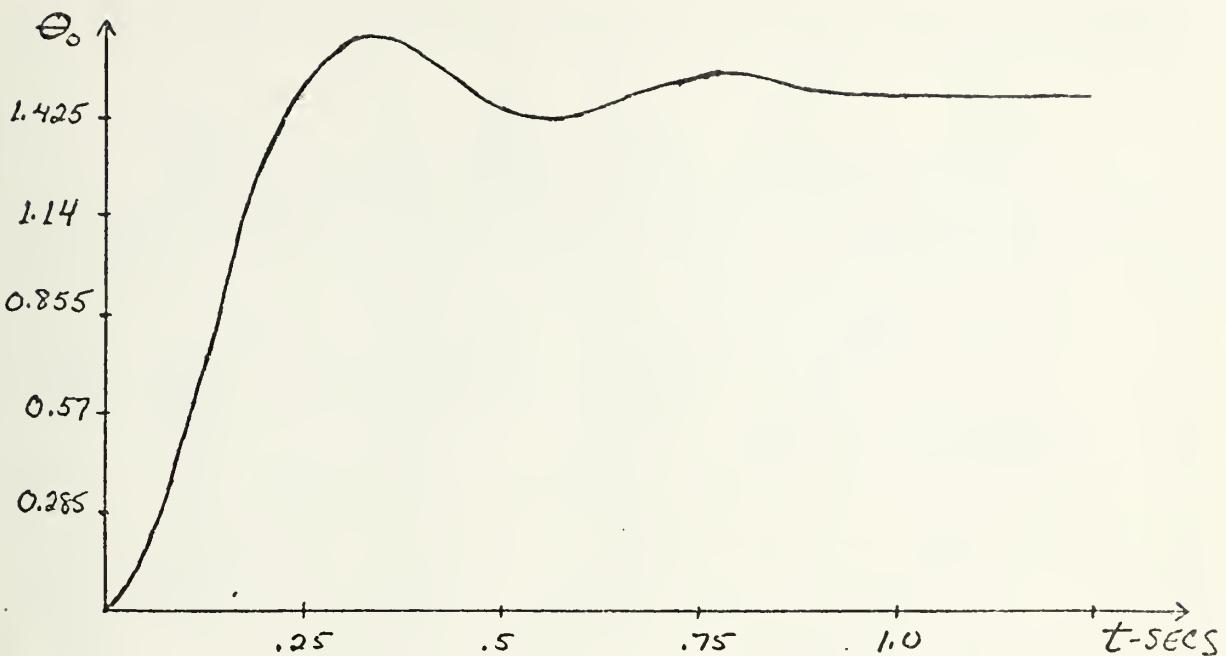


Figure 5.3
STEP RESPONSE FOR POINT A

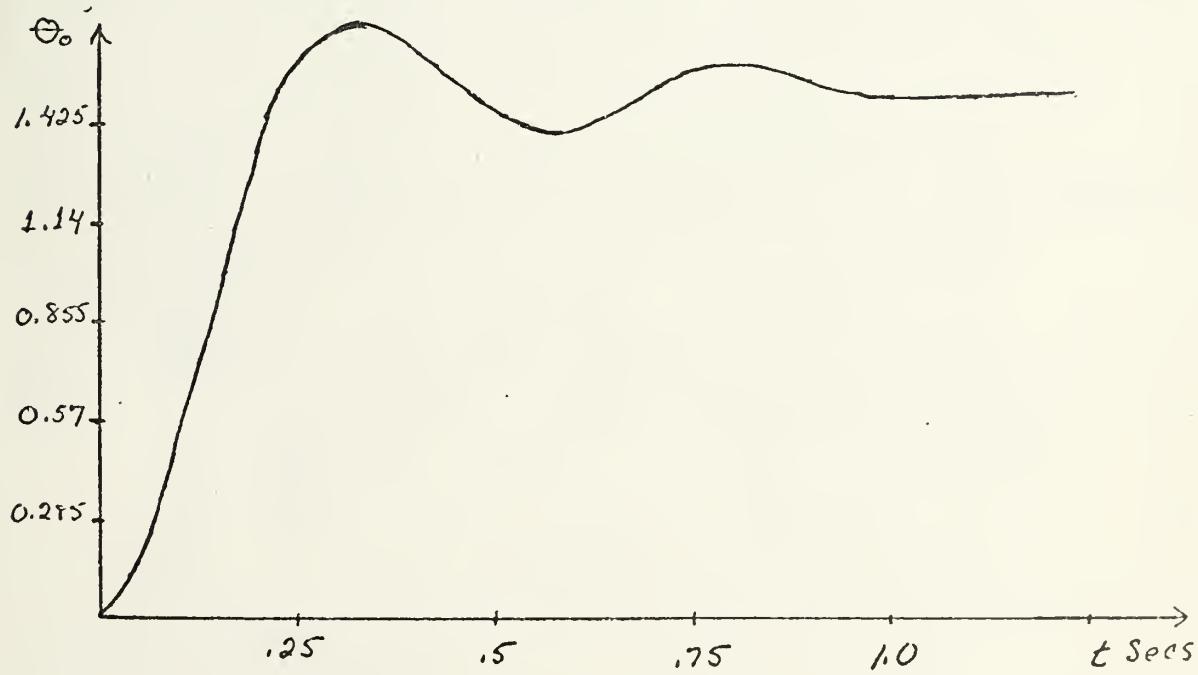
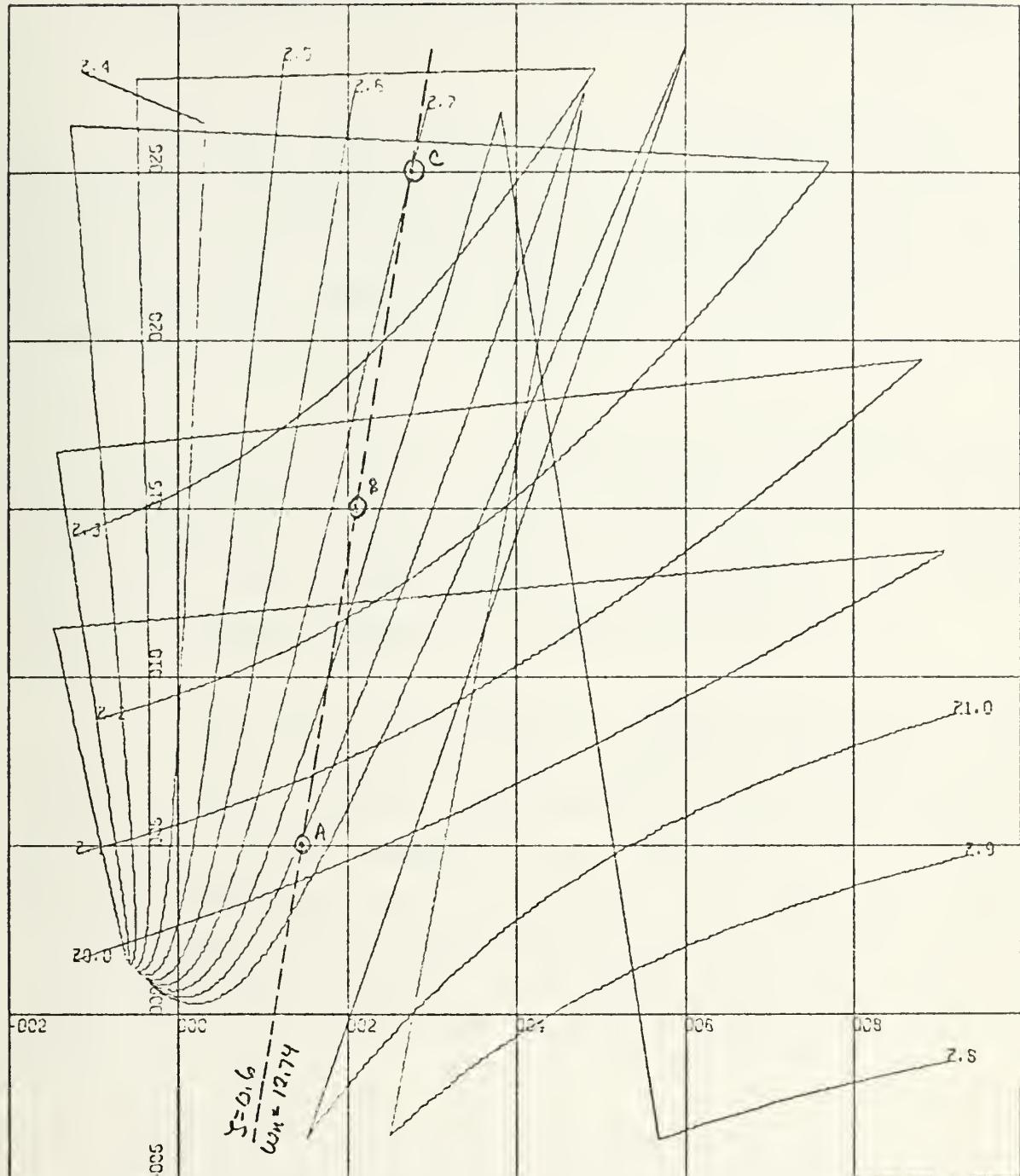


Figure 5.4
STEP RESPONSE FOR POINT B



α -scale: 20 units/inch
 β -scale: 50 units/inch

Figure 5.5

AIRCRAFT PITCH STABILIZATION SYSTEM PARAMETER PLANE

VI. CONCLUSIONS AND RECOMMENDATIONS FOR FURTHER STUDY

It has been shown that some of the original concepts of the singular line were in error due to previously unknown properties of the singular line and its relationship to the parameter plane. The main reason these properties were not found earlier was due mainly to standard computer programming practices and limitations.

Although it has been found that some limitations exist when using singular line concepts it has been shown that the basic concepts of singular lines can be used by the design engineer to easily obtain desired system response when more than one variable parameter is present in the system.

An example has been presented which shows that the singular line concept is an excellent method to obtain desired system response. The method will allow the designer to select either dominant root locations or, if it is desired, nondominant root locations. The root locations when selected from the singular line region of the parameter plane will remain fixed in their location as long as the alpha and beta parameters are restricted to the singular line.

For further study it is highly recommended that the case where true singular lines exist such as in example V be thoroughly studied to obtain a full understanding of why the singular point does not occur at infinity in the parameter plane. Once this is accomplished devise a method to force this type of condition at a desired zeta-omega value. If this can be accomplished it would allow the designer to

select a root location in the s-plane and use singular line concepts for easy design and compensation.

APPENDIX A

PARAM A

This is a modified version of the PARAM A program written by Nutting [5]. Originally written in FORTRAN for the CDC 1604 Computer, it has been modified to FORTRAN IV for the IBM 360/67 at Naval Postgraduate School. It has also been modified to allow the user to select the size of the output plot and to plot singular lines in the parameter plane.

However PARAM A will not determine the existence of singular lines. It is necessary for the user to know the value of ζ and ω_n of a singular line before this option can be used. These values can best be determined by use of the program SINGULAR LINE. The user then can either put these values into PARAM A or use the values to draw the singular line on the PARAM A output plot.

APPENDIX B

SINGULAR LINE

The purpose of the program SINGULAR LINE is to determine the existance of singular lines in the parameter plane. The program is written in FORTRAN IV for the IBM 360/67 computer and the CALCOMP plotter at Naval Postgrauate School.

The values read into the program are the coefficients of the characteristic equation, the values of zeta to be examined, and plotting information. The program contains three subroutines which are located in the source library of the IBM 360/67. These are POLRT which has the ability to solve up to a 36 order polynomial, DRAW which outputs the plot to the CALCOMP plotter, and BCNV which converts numeric symbols to alpha-numeric for use in DRAW.

The program takes the values of zeta and first generates the Chebyshev function of the second kind. It then generates the coefficients for the omega terms in the polynomial expansion of $B_{12}^C - B_{21}^C = 0$. This generation is accomplished by using the coefficients of the alpha and beta terms in the characteristic equation, the value of zeta and the Chebyshev function. The resulting omega coefficients are now input into the subroutine POLRT.

POLRT solves for the roots of the omega polynomial and outputs the results to the program. These roots are now examined for positive real values. If positive real roots are present there is a possibility of singular lines

existing for that value of zeta. If there are not positive real roots, there are no singular lines. This information is printed out and the next value of zeta is examined.

For the positive real roots B_1' , B_2' , C_1' , C_2' , D_1' , and D_2' of equation (2-27) are formed and used to insure that a singular line does exist. If a singular line does exist then B_1' , C_1' , and D_1' are used to generate the points along the line by using equation (2-28). These points are now input into the subroutine DRAW and are plotted and labeled.

The printed output includes the value of zeta and omega, the alpha and beta intercept points, the slope, and the label. If there are less than two points generated within the area of the plot there will be no plot of that line. This information is indicated on the printed output and the user can either change the plot scales or use the output line information to construct a singular line plot.

This procedure is repeated for each value of zeta, plotting all the lines on a single plot. Multiple runs for different characteristic equations are possible by using additional data decks. Each equation will have its own output plot.

SYSSIN DD DATA

SINGULAR LINE PROGRAM
AUGUST 1975
LT PAUL F. VAN TASSEL, USN

THIS PROGRAM IS A DESIGN AID FOR DETERMINING THE EXISTANCE OF SINGULAR LINES FOR A GIVEN VALUE OF ZETA FOR A SYSTEM WHOSE CHARACTERISTIC EQUATION HAS COEFFICIENTS $(B^2\alpha\beta + C^2\beta^2)$ WHERE β , C AND α ARE CONSTANTS AND $\alpha\beta$ AND β^2 ARE VARIABLE PARAMETERS. ZETA, THE ALPHAS AND THE GIVEN VALUES OF ZETA, THE PROGRAM WILL SOLVE FOR OMEGA AND THE ALPHAS AND β IN THE INTERCEPT POINTS OF THE SINGULAR LINE ON THE ALPHAS-BETA PLANE.

THE PROGRAM WILL PLOT ON A GRAPH, JP TO 9 INCHES BY 15 INCHES, SINGULAR LINES FOR A GIVEN VALUE OF ZETA BETWEEN ZERO AND ONE. IN THE EVENT A SINGULAR LINE IS NOT GRAPHED, A PRINTOUT OF THAT LINE WILL INCLUDE THE ALPHAS AND β . THE USER IS ALLOWED TO EITHER CHANGE THE PLOT SCALES OR TO PLOT THE LINE SEPARATELY.

THE FOLLOWING SYMBOLS ARE PERTINENT TO THE PROGRAM AND ARE USED AS INPUT DATA.

NUM THE NUMBER OF CHARACTERISTIC EQUATIONS TO BE EXAMINED.

NO THE ORDER OF THE CHARACTERISTIC EQUATION TO BE EXAMINED.
NZ THE NUMBER OF VALUES OF ZETA TO BE CONSIDERED.
(NOT TO EXCEED 25).
IW THE WIDTH OF THE OUTPUT GRAPH IN INCHES.
(NOT EXCEEDED 9 INCHES)
IH THE HEIGHT OF THE OUTPUT GRAPH IN INCHES.
(NOT EXCEEDED 15 INCHES)
IXUP THE DISTANCE IN INCHES OF THE ALPHA-AXIS FROM THE BOTTOM OF THE GRAPH.
IYRT THE DISTANCE IN INCHES OF THE BETA-AXIS FROM THE LEFT EDGE OF THE GRAPH.
IG DETERMINES IF THE GRAPH WILL HAVE A ONE INCH GRID.
IPNT PRINT OUT ALL PLOTTED POINTS. IF DESIRED

0010
0020
0030
0040
0050
0060
0070
0080
0090
0100
0110
0120
0130
0140
0150
0160
0170
0180
0190
0200
0210
0220
0230
0240
0250
0260
0270
0280
0290
0300
0310
0320
0330
0340
0350
0360
0370
0380
0390
0400
0410
0420
0430
0440
0450
0460
0470


```

* * * * * SET IPNT=0, IF NOT SET IPNT=1.
* * * * * ITITLE(1-5) THE FIRST TITLE LINE ON THE OUTPUT GRAPH.
* * * * * MAY CONTAIN UP TO 48 ALPHANUMERIC CHARACTERS (INCLUDING BLANKS).
* * * * * D(N) THE CONSTANT COEFFICIENTS IN THE CHARACTERISTIC EQUATION.
* * * * * READ IN ASCENDING ORDER.
* * * * * B(N) THE ALPHA TERM COEFFICIENTS. READ IN ASCENDING ORDER.
* * * * * C(N) THE BETA TERM COEFFICIENTS. READ IN ASCENDING ORDER.
* * * * * ZETA(N) THE VALUES OF ZETA FOR WHICH SINGULAR LINE-INTERPOLATION IS DESIRED. NUMBER MUST CORRESPOND WITH NZ.
* * * * * XSCALE THE DESIRED SCALE OF THE ALPHA-AXIS. MUST BE ONE SIGNIFICANT FIGURE.
* * * * * YSCALE THE DESIRED SCALE OF THE BETA-AXIS. MUST BE ONE SIGNIFICANT FIGURE.
* * * * * THE INPUT DATA FOR THIS PROGRAM IS INSERTED IN THE FOLLOWING MANNER.
* * * * * CARD 1 NUM - 13 FORMAT
* * * * * CARD 2 NZ,IW,IH,IYUP,IYRT,IS,IPNT - 8110 FORMAT
* * * * * CARD 3 ITITLE(1-6) - 6A8 FORMAT
* * * * * CARD 4 ITITLE(7-12) - 6A8 FORMAT
* * * * * CARD 5 Z(N) - 8E10.5 FORMAT
* * * * * CARD 6 B(N) - 8E10.5 FORMAT
* * * * * CARD 7 C(N) - 8E10.5 FORMAT
* * * * * CARD 8 ZETAN(N) - 8E10.5 FORMAT
* * * * * CARD 9 XSCALE - E10.5 FORMAT
* * * * * CARD 10 YSCALE - E10.5 FORMAT
* * * * * NOTE: CARDS 5 THROUGH 8 IT MAY BE NECESSARY TO USE MORE THAN ONE CARD, DEPENDING ON THE ORDER OF THE SYSTEM AND THE NUMBER OF ZETAS TO BE EXAMINED. WHEN THE NUMBER OF CHARACTERISTIC EQUATIONS (NZ) IS GREATER THAN ONE CARDS 2-10 ARE REPEATED.

```



```

C ****
C CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
REAL *4 DUM/1./,LABEL/,/
DIMENSION B(25),C(25),D(25),IER(4),ZETA(25),WNS(25),UZ(25),WN(25),
        1B1(25),C1(25),D1(25),G(2),B2(25),D2(25),WWX(25)
CALL ERSET(207,256,-1,0,209)

C READ IN THE NUMBER OF DATA SETS.
READ (5,28) NJM
DO 24 L0NG=1,NUM
WRITE (6,27)
READ (5,39) NO,VZ,IW,IH,IXUP,IYRT,IG,IPNT
NC = NO+1

C INITIALIZE ALL ARRAYS TO ZERO
DO 1 I=1,25
B(I) = 0.0
C(I) = 0.0
D(I) = 0.0
ZETA(I) = 0.0
WN(I) = 0.0
B1(I) = 0.0
C1(I) = 0.0
D1(I) = 0.0
B2(I) = 0.0
C2(I) = 0.0
D2(I) = 0.0
CONTINUE
1 CONTINUE

C READ AND WRITE THE TITLE FOR THE PLOT
READ (5,38) ITITLE(I),I=1,6
READ (5,38) ITITLE(I),I=7,12
WRITE (6,40) ITITLE(I),I=1,6
WRITE (6,40) ITITLE(I),I=7,12
READ AND WRITE THE CONSTANT COEFFICIENTS
WRITE (6,29)
READ (5,32) (D(N),N=1,NC)
WRITE (6,33) (D(N),N=1,NC)

C READ AND WRITE THE A-COEFFICIENTS
WRITE (6,30)
READ (5,32) (B(N),N=1,NC)

```



```

C READ AND WRITE THE B-COEFFICIENTS
C WRITE (6,33) {B(N),N=1,NC)
C WRITE (6,31) {C(N),N=1,NC)
C READ (5,32) (C(N),N=1,NC)
C WRITE (6,33) {C(N),N=1,NC)

C READ AND WRITE THE VALUES OF ZETA FOR SINGULAR LINES.
C WRITE (6,37)
C READ (5,32) (ZETA(N),N=1,NZ)
C WRITE (6,33) {ZETA(N),N=1,NZ}

C READ AND WRITE THE X-AXIS DIVISION (XSCALE/INCH)
C READ (5,34) XSCALE
C WRITE (6,35) XSCALE

C READ AND WRITE THE Y-AXIS DIVISION (YSCALE/INCH)
C READ (5,34) YSCALE
C WRITE (6,36) YSCALE
C WRITE (6,43)
DO 23 KK=1,NZ

C GENERATE CHEBYSHEV FUNCTIONS OF THE SECOND KIND FOR A GIVEN ZETA.
UZ(1) = -1.0
UZ(2) = 0.0
NN = NO+2
DO 2 M=3,NN
UZ(M) = 2.0*ZETA(KK)*UZ(M-1)-UZ(M-2)
2 CONTINUE

C GENERATE THE DMEGA(N) COEFFICIENTS FOR THE DETERMINATE OF THE
B1C2 - B2C1 MATRIX.
C
DO 3 I=1,25
WNS(I) = 0.0
WWX(I) = 0.0
3 CONTINUE
DO 4 I=1,NO
JJ = NC-I
DO 4 J=1,JJ
J2 = (2*I)+(J-2)
WNS(J2) = (B(I)*C(I+J)-C(I)*B(J+I))*((-1.)**(J+1.))*UZ(J+2)+WNS(J2)
4 CONTINUE
IF (WNS(J2).NE.0.0) JOD=J2
DO 5 I=1,NTWO
J = I+1
5 WWX(J) = WNS(I)

```



```

C      WWX(1) = 0.0
C      COMPUTE THE ROOT FOR THE EXPANSION OF (B1C2 - B2C1).
C      CALL POLKT (WWX,XWWX,JDD,RR,RI,IER)
C
C      I11 = 1
C      JDO = JDO+1
C      DO 6 I1=1,JDO
C
C      TEST TO DETERMINE THE EXISTANCE OF REAL ROOTS.
C      IF (RR(I1)*LE(0.0).OR.RI(I1)*NE.0.0) GO TO 6
C
C      I11 = I11+1
C      CONTINUE
C      I11 = I11-1
C      JT = 0
C      MC = 0
C      IF (WN(1)*NE.0.0) GO TO 7
C      WRITE(6,42) ZETA(KK)
C      GO TO 23
C
C      DO 22 K1=1,I11
C      IF (JT.LT.2.AND.MC.EQ.1) GO TO 8
C      MC = 2
C      IF (KK.EQ.1.AND.K1.EQ.1) MC=1
C      IF (KK.EQ.NZ.AND.K1.EQ.I1) MC=3
C      IF ((NZ.EQ.1.AND.I1.EQ.1) MC=0
C
C      CONTINUE
C
C      JT = 0
C      MC = 0
C      IF (WN(1)*NE.0.0) GO TO 7
C      GO TO 23
C
C      6    IF (JT.LT.2.AND.MC.EQ.1) GO TO 8
C      MC = 2
C      IF (KK.EQ.1.AND.K1.EQ.1) MC=1
C      IF (KK.EQ.NZ.AND.K1.EQ.I1) MC=3
C      IF ((NZ.EQ.1.AND.I1.EQ.1) MC=0
C
C      CONTINUE
C
C      GENERATE B1,C1, AND D1 FOR EACH VALUE OF OMEGA AT A GIVEN ZETA
C
C      DO 10 K2=1,NC
C      K3 = K2+1
C      IF ((K2-1)*NE.0) GO TO 9
C      B1(K1) = -B(K2)
C      C1(K1) = -C(K2)
C      D1(K1) = 0.0
C      B2(K1) = 0.0
C      C2(K1) = 0.0
C      D2(K1) = 0.0
C      GO TO 10
C
C      7    CONSQ = (-1)**(K2-1)
C      WNQN = WN(K1)**(K2-1)
C      WNQN = WNQ*WNQN
C      B1(K1) = B(K2)*UZ(K2)*WNQN+B1(K1)
C      D1(K1) = C(K2)*UZ(K2)*WNQN+C1(K1)
C      C1(K1) = D(K2)*UZ(K2)*WNQN+D1(K1)
C      D1(K1) = B(K2)*UZ(K3)*WNQN+B2(K1)
C      B2(K1) = C(K2)*UZ(K3)*WNQN+C2(K1)
C      C2(K1) = D(K2)*UZ(K3)*WNQN+D2(K1)
C
C      8    CONTINUE
C
C      9    CONSQ = (-1)**(K2-1)
C
C      10   CONTINUE

```



```

C TEST TO SEE IF SYSTEM OF EQUATIONS IS INDETERMINATE (B1=B2=C1=C2).
C
IF (B1(K1).NE.B2(K1)) GO TO 11
IF (B1(K1).NE.C1(K1)) GO TO 11
IF (B1(K1).NE.C2(K1)) GO TO 11
WRITE (6,44) ZETA(KK)
GO TO 24
CONTINUE
11 WRITE (6,42) ZETA(KK)
IDCK = 0
IF (C1(K1).NE.0.0 AND C2(K1).NE.0.0) GO TO 12
GO TO 24
12 IF (B1(K1).EQ.0.0.GR.B2(K1).EQ.0.0) ISCK=1
IF (B1(K1).EQ.0.0.GR.D2(K1).EQ.0.0) IDCK=1
IF (B2(K1).EQ.0.AND.IDCK.EQ.0) GO TO 14
IF (B2(K1).NE.B1(K1)) GO TO 13
IF (D2(K1).NE.D1(K1)) GO TO 13
GO TO 14
13 WRITE (6,42) ZETA(KK)
GO TO 24
14 CONTINUE
C AMIN = (IYINT-IXINT)*XSCALE
AMAX = IYSCALE*(0.25-IXUP)
BMIN = YSCALE*(IY-IXUP-0.25)
BMAX = (YMAX-AMIN)/300.
DELA = DELA
SLOPE = -B1(K1)/C1(K1)
YINT = -(YINT/SLOPE)
XINT = 0
AG(1) = 0.0
AG(2) = 0.0
BG(1) = 0.0
BG(2) = 0.0
C COMPUTE THE X(ALPHA) AND Y(BETA) INTERCEPTS.
C
15 AG(1) = AMIN
BG(1) = -(DI(K1)/C1(K1))-(B1(K1)/C1(K1))*AG(1)
JT = 1
IF (B1(K1).GE.3MIN.AND.BG(1).LE.BMAX) GO TO 19
IF (AG(1)*GT.4MAX) GO TO 19
AG(1) = AG(1)+DELA
GO TO 15
16 AG(2) = AMAX*(AMAX-XINT)
BG(2) = SLOPE*(AMAX-XINT)

```



```

1 IF (BG(2).LE.BMAX.AND.BG(2).GE.BMIN) GO TO 18
2 IF (SLOPE.LT.0.) GO TO 17
3 BG(2) = BMAX
4 AG(2) = (BMAX-YINT)/SLOPE
5 GO TO 18
6
7 BG(2) = BMIN
8 AG(2) = (BMIN-YINT)/SLOPE
9 CONTINUE
10 LABZ(K1) = (K1*10)+KK
11
12 C CONVERT THE LABELS FROM NUMERIC TO ALPHA-NUMERIC CHARACTERS.
13 CALL BCNV(LABZ(K1),LABBZ,10,4)
14 WRITE(6,41) ZETA(KK),WN(K1),XINT,YINT,SLOPE,LABZ(K1)
15
16 PRINT OUT THE POINTS WHICH ARE PLOTTED ON THE GRAPH.
17 IF (IPNT.NE.0) GO TO 21
18 WRITE(6,25)
19 DO 20 I4=1,14
20 14=1.25*AG(14),BG(14)
21 WRITE(6,43)
22 CONTINUE
23
24 C PLOT THE SINGULAR LINES ON THE GRAPH.
25 CALL DRAW(JT,BG,MC,O,LABBZ,ITITLE,XSCALE,YSCALE,IXUP,IYRT,2,2,
26 1,IWTHIG,LAST)
27 CONTINUE
28 CONTINUE
29 CONTINUE
30 CONTINUE
31 FORMAT(4X,'POINTS OF THIS SINGULAR LINE WHICH ARE PLOTTED.',//,4X
32 1,'ALPHA',4X,BETA,/,1)
33 FORMAT(4X,E12.5,4X,E12.5)
34 FORMAT(4X,13) THE INPUT DATA IS:',//,1)
35 FORMAT(4X,'CONSTANT COEFFICIENTS IN ASCENDING ORDER',//)
36 FORMAT(4X,'ALPHA COEFFICIENTS IN ASCENDING ORDER',//)
37 FORMAT(4X,'BETA COEFFICIENTS IN ASCENDING ORDER',//)
38 FORMAT(8E10.5)
39 FORMAT(4X,8E14.5)
40 FORMAT(5E10.5)
41 FORMAT(4X,'SCALE:',F10.5,' UNITS PER INCH',//)
42 FORMAT(4X,'SCALE:',F10.5,' UNITS PER INCH',//)
43 FORMAT(4X,'4X,',YSCALE,'4X,',ZETA VALUES FOR SINGULAR LINES',//)
44 FORMAT(6A8)
45 FORMAT(8I10)
46 FORMAT(4X,6A3)
47 FORMAT(4X,E12.5,4X,E12.5,6X,E12.5,9X,E12.5,6X,E12.5,6X,E12.5,14,/)

```



```

42 FORMAT (4X,E12.5,5X,'NO SINGULAR LINES EXIST FOR THIS VALUE OF ZET' SL 3360
42 1A,/,)
43 FORMAT ('1 SINGULAR LINE DATA'//,9X,'ZETA',12X,'OMEGA',6X,'X' SL 3370
43 1A,/,)
44 FORMAT ('1 AXIS INTERCEPT',9X,'SLOPE',9X,'LABEL',/,)'/' SL 3380
44 1A,/,)
45 FORMAT ('1 SYSTEM OF EQUATIONS IS INDETERMINATE.' RANK = SL 3390
45 1A,/,)
46 END SL 3400
46 1A,/,)

```


CARD 16. (8E10.5 FORMAT)
 ONEGAN STARTING VALUE (E10.5 FORMAT), USE BLANK CARD
 IF NZ=0)

CARD 17. IXSCALE AND YSCALE {2E10.5 FORMAT}
 CARD 18. IHIGH {2I5 FORMAT} **** SPECIFICATIONS:
 DATA SUBMISSION FOR THIRD PARAMETER SPECIFICATION
 CARD 0. SAME AS CARD 0 IN PREVIOUS SECTION
 CARD 1. SAME AS CARD 1 IN PREVIOUS SECTION EXCEPT NE NOW
 CARD 2. SAME AS CARD 2 IN PREVIOUS SECTIONS OF THE THIRD PARAMETER
 CARD 3. SAME AS CARD 3 IN PREVIOUS SECTIONS OF CARD 6 OF PREVIOUS
 CARD 4. SAME AS CARD 4 IN PREVIOUS SECTIONS OF CARD 6 OF PREVIOUS
 CARD 5. THRU CARD 8 SAME AS CARD 5 THRU CARD 6 OF PREVIOUS
 CARD 5. THRU CARD 8 SAME AS CARD 5 THRU CARD 6 OF PREVIOUS
 NE**NZ FOR LABZ
 NE**NW FOR LABZW
 NE**NW FOR LABZ
 LABELS AND VALUES ARE READ FOR ALL VALUES OF NE FOR
 EACH LABEL 12 AND VALUE OF THE CURVE DESIGNATED
 SECTION 12 IN PREVIOUS SECTION
 SECTION 13 IN PREVIOUS SECTION
 SECTION 14 IN PREVIOUS SECTION
 SECTION 15 IN PREVIOUS SECTION
 SECTION 16 IN PREVIOUS SECTION
 SECTION 17 IN PREVIOUS SECTION
 SECTION 18 IN PREVIOUS SECTION
 SECTION 19 IN PREVIOUS SECTION
 SECTION 20 IN PREVIOUS SECTION
 SECTION 21 IN PREVIOUS SECTION
 SECTION 22 IN PREVIOUS SECTION
 SECTION 23 IN PREVIOUS SECTION
 SECTION 24 IN PREVIOUS SECTION
 SECTION 25 IN PREVIOUS SECTION
 SECTION 26 IN PREVIOUS SECTION
 SECTION 27 IN PREVIOUS SECTION
 SECTION 28 IN PREVIOUS SECTION

CARD 9. THRU CARD 12 SAME AS CARD 9 THRU CARD 12 IN PREVIOUS
 SECTION 12 IN PREVIOUS SECTION
 SECTION 13 IN PREVIOUS SECTION
 SECTION 14 IN PREVIOUS SECTION
 SECTION 15 IN PREVIOUS SECTION
 SECTION 16 IN PREVIOUS SECTION
 SECTION 17 IN PREVIOUS SECTION
 SECTION 18 IN PREVIOUS SECTION
 SECTION 19 IN PREVIOUS SECTION
 SECTION 20 IN PREVIOUS SECTION
 SECTION 21 IN PREVIOUS SECTION
 SECTION 22 IN PREVIOUS SECTION
 SECTION 23 IN PREVIOUS SECTION
 SECTION 24 IN PREVIOUS SECTION
 SECTION 25 IN PREVIOUS SECTION
 SECTION 26 IN PREVIOUS SECTION
 SECTION 27 IN PREVIOUS SECTION
 SECTION 28 IN PREVIOUS SECTION

CARD 13. SAME AS CARD 13 OF THIRD PARAMETER (E10.5 FORMAT)
 CARD 14. SAME AS CARD 14 OF THIRD PARAMETER (E10.5 FORMAT)
 CARD 15. SAME AS CARD 15 OF THIRD PARAMETER (E10.5 FORMAT)
 CARD 16. SAME AS CARD 16 OF THIRD PARAMETER (E10.5 FORMAT)
 THE COEFFICIENTS OF THE CHARACTERISTIC EQUATION MUST BE ENTERED IN
 SUBROUTINE COEF. THE THIRD PARAMETER E MAY APPEAR LINEARLY OR
 NON-LINEARILY IN ANY OF THE COEFFICIENTS. FOR EXAMPLE, GIVEN THE
 CHARACTERISTIC EQUATION:

$$(BJ^{(4)} * ALFA + CJ^{(4)} * BETA + DJ^{(4)} * S)^2$$

$$+ (BJ^{(3)} * ALFA + CJ^{(3)} * BETA + DJ^{(3)} * S) * S$$

$$+ (BJ^{(2)} * ALFA + CJ^{(2)} * BETA + DJ^{(2)} * S) * S^2$$

$$+ (BJ^{(1)} * ALFA + CJ^{(1)} * BETA + DJ^{(1)} * S) * S^3$$

$$+ (BJ^{(0)} * ALFA + CJ^{(0)} * BETA + DJ^{(0)} * S) * S^4 = 0$$
 THE COEFFICIENTS OF THE ABOVE EQUATION CAN BE ENTERED IN COEF AS
 FOLLOWS:
 COMMON E, BJ, CJ, DJ
 BJ(4)=1.
 CJ(4)=0.
 DJ(4)=11.
 BJ(3)=11.
 CJ(3)=E
 DJ(3)=10.
 BJ(2)=10.
 CJ(2)=28.
 BJ(1)=28.
 CJ(0)=6


```

      WRITE (6,116) ((LABW(M,N),M=1,NW),N=1,NEE)
      WRITE (5,114) ((LABW(M,N),M=1,NW),N=1,NEE)
      WRITE (6,117) ((LABZ(W(M,N),M=1,NZ_W),N=1,NEE),
      WRITE (5,113) ((LABZ(W(M,N),M=1,NZ_W),N=1,NEE),
      WRITE (6,114) ((ZETA(M),M=1,NZ),
      WRITE (5,118) ((ZETA(M),M=1,NZ),
      WRITE (6,119) ((SIGMA(M),M=1,NS),
      WRITE (5,118) ((SIGMA(M),M=1,NS),
      WRITE (6,121) ((W(M),M=1,NW),
      WRITE (5,118) ((W(M),M=1,NW),
      WRITE (6,122) ((W(M),M=1,NW),
      WRITE (5,118) ((W(M),M=1,NW),
      WRITE (6,123) ((ZW(M),M=1,NW),
      WRITE (5,118) ((ZW(M),M=1,NZ_W),
      WRITE (6,124) ((ZW(M),M=1,NZ_W),
      WRITE (6,124) ((D(J(N),N=1,NC),
      WRITE (5,118) ((D(J(N),N=1,NC),
      WRITE (6,119) ((BJ(N),N=1,NC),
      WRITE (5,118) ((BJ(N),N=1,NC),
      WRITE (6,125) ((BJ(N),N=1,NC),
      WRITE (5,119) ((BJ(N),N=1,NC),
      WRITE (6,126) ((CJ(N),N=1,NC),
      WRITE (5,118) ((CJ(N),N=1,NC),
      WRITE (6,119) ((CJ(N),N=1,NC),
      GO TO 6
      WRITE (5,127) ((EJ(N),N=1,NE)
      WRITE (5,118) ((EJ(N),N=1,NE)
      CONTINUE
      READ (5,128) WN
      READ (5,129) XSCALE, YSCALE
      READ (5,130) WN
      READ (5,131) XSCALE, YSCALE
      READ (5,132) XSCALE, YSCALE
      READ (5,143) IWIDE, IHIGH
      READ (5,144) IWIDE, IHIGH
      ROG = AIYXUP-O.25
      DAV = -ROG*XSCALE
      ADAV = -DAV*YSCALE
      ROGE = IWIDE-O.25-AIYRIT
      DAVE = IHIGH-O.25-AIXUP

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AROGE = ROGE**XSCALE
ADAVE = DAVE**YSCALE
FRAN = AROG
CHEK = ADAV
IF (NZ) 18,18,7
      GO TO (8,9,10,11,12,13,14,15,16,17), ND
      8   G=1.0076
      GO TO 18
      9   G=1.016
      10  G=1.0245
      11  G=1.0312
      12  G=1.0394
      13  G=1.0483
      14  G=1.0568
      15  G=1.0633
      16  G=1.071
      17  G=1.078
      18  CONTINUE
      19  AG(1)=0.0
      20  AG(2)=0.0
      21  AG(1)=XSCALE
      22  AG(2)=XSCALE
      23  AG(1)=0.0
      24  AG(2)=0.0
      25  CALL DRAW(2,4G,BG,1,0,LABEL,ITITLE,XSCALE,YSCALE,1)
      26  CALL CUEF(NE,23,23,22)
      27  IF (IPRINT) 21,21,20
      28  WRITE (6,133)
      29  WRITE (6,134)
      30  MOD = 2
      31  IF (NZ) 38,38,19
      32  IF (IPRINT) 21,21,20
      33  WRITE (6,133)
      34  WRITE (6,134)
      35  MOD = 2
      36  M=1,NZ
      37  DO 37 M=1,NEE
      38  E=J(ME)
      39  IF (NE) 23,23,22
      40  CALL CUEF
      41  DO 36 M=1,NZ
      42  J=0
      43  WNA = WN
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      781 IF (NE) 23,23,22
      782 CALL CUEF
      783 DO 36 M=1,NZ
      784 J=0
      785 WNA = WN
      786 DO 37 M=1,NEE
      787 E=J(ME)
      788 IF (NE) 23,23,22
      789 CALL CUEF
      790 DO 36 M=1,NZ
      791 J=0
      792 WNA = WN
      793 DO 37 M=1,NEE
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      795 IF (NE) 23,23,22
      796 CALL CUEF
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      800 DO 37 M=1,NEE
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      810 CALL CUEF
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      816 IF (NE) 23,23,22
      817 CALL CUEF
      818 DO 36 M=1,NZ
      819 J=0
      820 WNA = WN
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      822 E=J(ME)
      823 IF (NE) 23,23,22
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      826 J=0
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      903 J=0
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      906 E=J(ME)
      907 IF (NE) 23,23,22
      908 CALL CUEF
      909 DO 36 M=1,NZ
      910 J=0
      911 WNA = WN
      912 DO 37 M=1,NEE
      913 E=J(ME)
      914 IF (NE) 23,23,22
      915 CALL CUEF
      916 DO 36 M=1,NZ
      917 J=0
      918 WNA = WN
      919 DO 37 M=1,NEE
      920 E=J(ME)
      921 IF (NE) 23,23,22
      922 CALL CUEF
      923 DO 36 M=1,NZ
      924 J=0
      925 WNA = WN
      926 DO 37 M=1,NEE
      927 E=J(ME)
      928 IF (NE) 23,23,22
      929 CALL CUEF
      930 DO
```



```

DO 35 L=1,300
D1 = 0.0
D2 = 0.0
C1 = 0.0
C2 = 0.0
B1 = 0.0
B2 = 0.0
C
DO 26 N=1,NC
K = (K,-1) 25,24,25
U = 0.0
24 U1 = -1.0
U2 = 2.0*ZETA(M)*U-U1
D1 = (-1.0)**K*D(J(N))*WNAA**K*U1+D1
D2 = (-1.0)**K*D(J(N))*WNAA**K*U+D2
C1 = (-1.0)**K*C(J(N))*WNAA**K*U1+C1
C2 = (-1.0)**K*C(J(N))*WNAA**K*U+C2
B1 = (-1.0)**K*3*J(N)*WNAA**K*U1+B1
B2 = (-1.0)**K*3*J(N)*WNAA**K*U+B2
U1 = U2
U =
C
Z = 1.0E-60
IF (ABS(B1*C2-B2*C1)-Z ) 27,27,28
GO TO 35
27 J = J+1
A(J) = (C1*D2-C2*D1)/(B1*C2-B2*C1)
B(J) = (B2*D1-B1*D2)/(B1*C2-B2*C1)
IF (INT(1.35)*A(J)) 29,29,30
WRITE(6,31) A(J),B(J),WNA,ZETA(M),E
30 IF ((RAN-A(J)) 31,31,35
31 IF ((A(J)-AR0GE) 32,32,35
32 IF ((CHEK-B(J)) 33,33,35
33 IF ((B(J)-ADAVE) 34,34,35
34 JG = J6+1
AG(JG) = A(J)
BG(J) = B(J)
WNA = G*WNA
C
CALL DRAW (JG,AG,BG,MOD,O,LABZ(M,ME),ITITLE,XSCALE,YSCALE,IYR,
1 ITET2,IWIDE,IHIGH,1,LAST)
36 CONTINUE
C
37 CONTINUE
C
38 IF (NS) 64,64,39
39 IF (IPRINT) 41,41,40

```



```

40 WRITE ( 6,136)
C   41 DO 65 ME=1,NEE
      E = EJ(ME)
      IF (NE.EQ.43,43,42
C   42 CALL COEF
C   43 DO 64 M=1,NS
      DD = 0.0
      CC = 0.0
      BB = 0.0
C   44 CONTINUE
C
      DO 44 N=1,NC
      K = N-1
      DUMMY5 = 0.0
      IF (SIGMA(M).NE.0.0) DUMMY5=SIGMA(M)**K
      DD = (-1.0)**K*D(J(N)*DUMMY5+DD
      CC = (-1.0)**K*C(J(N)*DUMMY5+CC
      BB = (-1.0)**K*B(J(N)*DUMMY5+BB
C
      J = 1
      A(J) = -DD/BB
      B(J) = 0.0
      IF ((IPRINT).EQ.46,46,45
      WRITE (6,138) A(J),B(J),SIGMA(M),E
45    IF (AROG-A(J)).LT.47,47,49
46    IF (A(J)-AROGE).LT.48,48,49
47    J = J+1
48
49    A(J) = 0.0
      B(J) = -DD/CC
      IF ((IPRINT).EQ.51,51,50
      WRITE (6,138) A(J),B(J),SIGMA(M),E
50
51    IF (ADAV-B(J)).LT.52,52,54
52    IF (B(J)-ADAVE).LT.53,53,54
53    J = J+1
54    B(J) = (HIGH-0.25-AIXUP)**YSCALE
      A(J) = (-CC*B(J)-DD)/BB
      IF ((IPRINT).EQ.56,56,55
      WRITE (6,138) A(J),B(J),SIGMA(M),E
55
56    IF (AROG-A(J)).LT.57,57,58
57    IF (A(J)-AROGE).LT.63,63,63,58
58    A(J) = (IWIDE-0.25-AIYRIT)*XSCALE
      B(J) = (-BB*B(J)-DD)/CC
      IF ((IPRINT).EQ.60,60,59
      WRITE (6,138) A(J),B(J),SIGMA(M),E
59
60    IF (ADAV-B(J)).LT.61,61,62

```



```

61 IF (B(J)-ADAVE) J-1 63,63,62
62 CALL DRAW (J,A,B,2,0,LABS(M,ME),ITITLE,XSCALE,YSCALE,IYRITE,2
63 I2,IWIDE,IHIGH,1,LAST)
64 CONTINUE
C   65 CONTINUE
C     IF (NZW) 84,84,66
C     IF ((IPRINT) 68,68,67
C       WRITE (6,139)
C       WRITE (6,140)
C   68 DO 83 M=1,NEE
C     E=EJ(NE)
C     IF (NE) 70,70,69
C   69 CALL COEF
C   70 DO 82 M=1,NZW
C     J=0
C     JG=0
C     AZETA = .00333
C   81 L=1,299
C     WN = ZW(M)/AZETA
C     D1 = 0.0
C     D2 = 0.0
C     C1 = 0.0
C     C2 = 0.0
C     B1 = 0.0
C     B2 = 0.0
C   82
C   83 DO 73 N=1,NC
C     K=N-1
C     IF (K) 72,71,72
C     Q1 = 0.0
C     Q2 = -1.0/WN**2
C     D2 = DJ(N)*Q1+D2
C     C2 = CJ(N)*Q1+C2
C     B2 = BJ(N)*Q1+B2
C     D1 = DJ(N)*Q+D1
C     C1 = CJ(N)*Q+C1
C     B1 = BJ(N)*Q+B1
C     Q2 = -2.0*ZN(M)*Q1-WN**2*Q
C     Q = Q1
C     Q1 = Q2
C   73
C     IF (ABS(B1*C2-B2*C1)-Z) 81,81,74

```



```

74 J = J+1
    A(J) = (C1*D2-C2*D1)/((B1*C2-B2*C1)
    B(J) = (B2*D1-B1*D2)/((B1*C2-B2*C1)
    IF (IPRINT) 76,75,A(J),B(J),ZW(M),E
    WRITE(6,138)A(J),B(J),ZW(M),E
    75 WRITE(FRAN-A(J)),77,77,81
    76 IF (A(J)-ARCGE) 78,78,81
    77 IF (CHEK-B(J)) 79,79,81
    78 IF (B(J)-ADAVE) 80,80,81
    79 IF (JG = JG+1)
    80 AG(JG) = A(J)
    BG(JG) = B(J)
    81 AZETA = AZETA+.00333
C     CALL DRAW (JG,AG,BG,2,0,LABZM(M,ME),ITITLE,XSCALE,YSCALE,IYUP,IYRI)
C     1 TE,2,2,IWIDE,IHIGH,i,LAST)
C     82 CONTINUE
C     83 CONTINUE
C     84 IF (NW) 103,103,85
    85 IF (IPRINT) 87,87,86
    86 WRITE(6,141)
    87 WRITE(6,142)
C     87 DO 102,ME=1,NEE
    E = EJ(ME)
    IF (NEE) 89,89,88
    88 CALL COEF
C     89 DO 101,M=1,NW
    J = 0
    JG = 0
    AZETA = 0.0
C     DO 100,L=1,300
    D1 = 0.0
    D2 = 0.0
    C1 = 0.0
    C2 = 0.0
    B1 = 0.0
    B2 = 0.0
C     DO 92,N=1,NC
    K = N-1
    IF (K) 91,90,91
    90 U = 0.0
    U1 = -1.0

```



```

91 U2 = 2.0*AZETA*U-U1
      D1 = (-1.0)**K*DJ(N)*W(M)**K*U1+D1
      D2 = (-1.0)**K*CJ(N)*W(M)**K*U+D2
      C1 = (-1.0)**K*CJ(N)*W(M)**K*U1+C1
      C2 = (-1.0)**K*3J(N)*W(M)**K*U1+C2
      B1 = (-1.0)**K*3J(N)*W(M)**K*U+B1
      B2 = (-1.0)**K*3J(N)*W(M)**K*U+B2
      U1 = U2
92 U = U2
C   IF (ABS(B1*C2-B2*C1)-Z) 100,100,93
93 J = J+1
      A(J) = (C1*D2-C2*D1)/(B1*C2-B2*C1)
      B(J) = (B2*D1-31*D2)/(B1*C2-B2*C1)
      IF (IPRINT) 95,95,94
      WRITE(6,135) A(J),B(J),W(M),AZETA,E
94 IF (FRAN-A(J)) 96,96,100
      IF (A(J)-AROUSE) 97,97,100
95 IF (CHEK-B(J)) 98,98,100
96 IF ((B(J))-ADAVE) 99,99,100
97 JG = JG+1
98 AG(JG) = A(J)
99 BG(JG) = B(J)
100 AZETA = AZETA+.00333
C   101 CALL DRAW (JG,AG,BG,MOD,0,LABW(M,ME),ITITLE,XSCALE,YSCALE,IXUP,IYR
      ITITLE,2,2,IWIDE,IHIGH,1,LAST)
C   102 CONTINUE
C   103 AG(1) = 0.0
      BG(1) = 0.0
      AG(2) = XSCALE
      BG(2) = 0.0
      CALL DRAW (2,AG,BG,3,0,LABEL,ITITLE,XSCALE,YSCALE,IXUP,IYRITE,2,2,
      1,IWIDE,IHIGH,1,LAST)
104 CONTINUE
C   STOP
C   105 FORMAT (1H1,17HTHE INPUT DATA IS,///)
      106 FORMAT ((6A8)
      107 FORMAT (3X,6A8)
      108 FORMAT (//,8X,2HN0,8X,2HNZ,8X,2HNS,8X,2HNW,7X,3HNZW,6X,4HI
      109 1XUP,3X,7HIYRIGHT,/,FORMAT ((8I10)
      110 FORMAT (//,4X,6HIPRINT,8X,2HNE,/,FORMAT (2I10)
      111 FORMAT (//,4X,6HIPRINT,8X,2HNE,/,FORMAT (2I10)

```



```

112 FORMAT (//,10X,' LABZ ',,/)
113 FORMAT (20A4)
114 FORMAT (3X,10X,*)
115 FORMAT (3X,10X,*)
116 FORMAT (3X,10X,*)
117 FORMAT (3X,10X,*)
118 FORMAT (3X,10X,*)
119 FORMAT (3X,8E10.5)
120 FORMAT (3X,10X,*)
121 FORMAT (3X,10X,*)
122 FORMAT (3X,10X,*)
123 FORMAT (3X,10X,*)
124 FORMAT (3X,10X,*)
125 FORMAT (3X,10X,*)
126 FORMAT (3X,10X,*)
127 FORMAT (3X,10X,*)
128 FORMAT (3X,10X,*)
129 FORMAT (2E10.5)
130 FORMAT (5X,E10.5,8X,E10.5,*)
131 FORMAT (5X,*SCALE*,10X,*)
132 FORMAT (4X,*PLOT WIDTH*,4X,*)
133 FORMAT (1H1,10X,*CONSTANT ZETA CURVES*)
134 FORMAT (15X,5HALPHA,16X,4HBETA,15X,5HOMEWA,16X,4HZETA,5X,15HTH1
1RD PARAMETER,/)
135 FORMAT (5E20.5)
136 FORMAT (1H1,10X,*CONSTANT SIGMA CURVES*
137 FORMAT (15X,5HALPHA,16X,4HBETA,15X,5HSIGMA,5X,15HTHIRD PARAMETER,/
138 FORMAT (4E20.5)
139 FORMAT (1H1,25HC CONSTANT ZETA-OMEGA CURVES*,/)
140 FORMAT (15X,5HALPHA,16X,4HBETA,10X,IGHZETA-OMEGA,5X,15HTHIRD PARAM
141 IETER,/)
142 FORMAT (1H1,21HC CONSTANT OMEGA CURVES*,/)
143 FORMAT (15X,5HALPHA,16X,4HBETA,15X,5HOMEWA,15X,5HAZETA,5X,15HTHIRD
144 FORMAT (215)
144 FORMAT (/,5X,11,* INCHES*,6X,12,* INCHES*,/)

SUBROUTINE COEF
DIMENSION BJ(100), CJ(100), DJ(100)
COMMON E,BJ,CJ,DJ
RETURN
END
PARA5280
PARA5290
PARA5300
PARA5310
PARA5320
PARA5330
PARA5340
PARA5350
PARA5360
PARA5370
PARA5380
PARA5390
PARA5400
PARA5410
PARA5420
PARA5430
PARA5440
PARA5450
PARA5460
PARA5470
PARA5480
PARA5490
PARA5500
PARA5510
PARA5520
PARA5530
PARA5540
PARA5550
PARA5560
PARA5570
PARA5580
PARA5590
PARA5600
PARA5610
PARA5620
PARA5630
PARA5640
PARA5650
COEF10
COEF20
COEF30
COEF40
COEF50

```


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